

Manufacturing and Condition-based Structural Health Monitoring of CFRPs

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Hanyang University ERICA



Present for: Anyone interested in composites
Updated on: July 18, 2025

01 INTRODUCTION



02 RESEARCH ACHIEVEMENTS

03 FUTURE PLAN

04 SUMMARY

05 APPENDIX

Credentials

Job

Assistant Professor	September, 2023 – Current	Department of Mechanical Engineering, Hanyang University ERICA
Senior Researcher	July, 2021 – August, 2023	Department of Composite Materials Research, Korea Institute of Materials and Science (KIMS)

Education

Post-doc. Researcher	September, 2020 – July, 2021	Mechanical Engineering, UNIST (Composite rapid manufacturing and AI)
Master's and Ph.D.- Combined Degree	March, 2014 – August, 2020	Mechanical Engineering, UNIST (Composite manufacturing, AI, and self-sensing)
Bachelor's Degree	March, 2010 – February, 2014	Mechanical Engineering & Electrical and Electronic Engineering, UNIST

Thesis Title (Ph.D.)

Characterization and Optimization of Self-sensing Performance of CFRP
Using Electrical Resistance Analysis and Electrical Circuit Models

Scholarship

Tuition and Book Fees	2010 Spring – 2019 Fall	34,739,000 won for Undergraduate Course and 97,562,000 won for M.S.-Ph.D. Combined Program
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Major Research Interests

Composite Manufacturing
(Rapid manufacturing, AI optimizing and
digitalized real-time manufacturing monitoring)

Composite Joining
(Fastener-free joining of CFRTPs and
self-assembly of sandwich composites)

**Smart Structural
Health Self-monitoring**
(Structural health monitoring,
prognostics and health management
and non-destructive evaluation)



조교수 노형도
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한양대학교 기계공학과 조교수 2023 ~ 현재
 한국재료연구원 선임연구원 2021 ~ 2023
 울산과학기술원 박사후연구원 2020 ~ 2021

- 연구실 정보
 - 연구실 위치 : ERICA 제5공학관 228호
 - 연구실 연락처 : 031-400-5246
- 기타 연구실 관련 정보
 - 현재 구성원 : 석사과정 5명, 석박통합과정 1명, 학부연구원 5명
 - 연구실 홈페이지 : <https://rhd1213.github.io/>

주요 연구 분야: 탄소섬유강화 플라스틱의 성형, 접합, 건전성 진단 및 수명 예측 with

Composite Design Manufacturing

Joining

Rapid manufacturing

Analyzing

Smart Composites Functionalizing

Specification, Circuit modeling, Reduced circuit

1. Elastic, 2. Indentation, 3. Delamination, 4. Puncture

Stress, Fracture, Deformation, Strain

Multi-function, Forecasting

Manufacturing

Carbon fiber reinforced plastic

Interflow interaction, Interply interaction, Tow failure

Radial shrinkage, Axial elongation

FLIR

Composites to Your Life

Lightweight

Artificial Intelligence

Structural battery

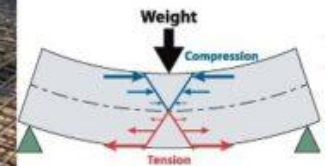
Internet of things

Deformation sensing

Vibrational energy

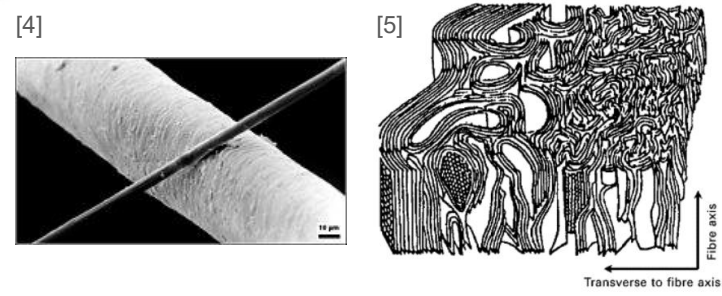
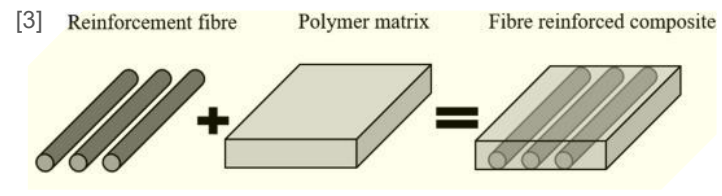
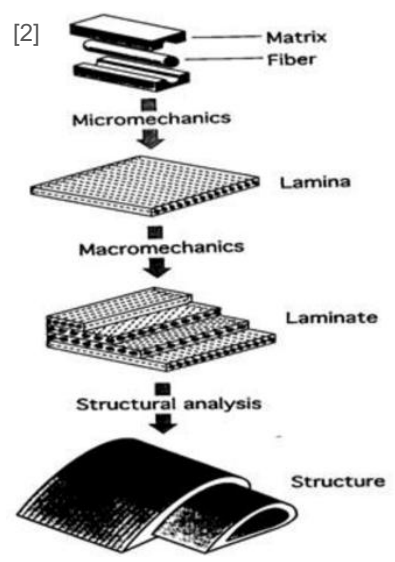
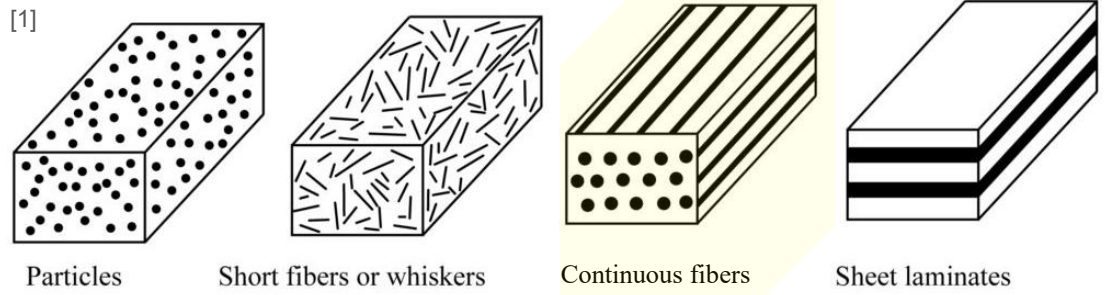
Composite

A composite material is a material which is produced from two or more constituent materials. (Wikipedia)

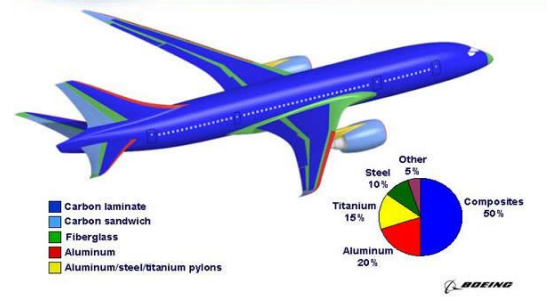


Introduction (Backgrounds)

► Composites – CFRP – Manufacturing & Safety – AI

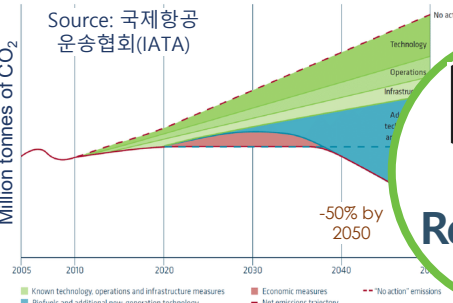


[7] **Composite Solutions Applied Throughout the 787**



Source: [1] Bartleby, [2] Progressive failure analysis of composite materials using the Puck failure criteria, [3] Modeling of reinforcement fibers and textiles, [4] Composite material –Wiki, [5] S. C. Bennett and D. J. Johnston. (1978), [6] BMW i3, [7] Boeing 787, [8] Aloha airlines

Mega-trend of Future Mobility

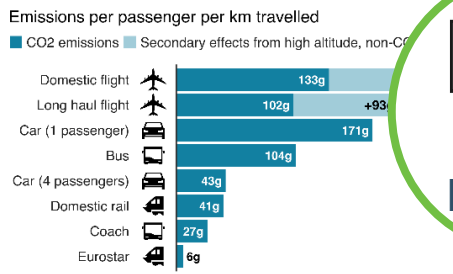


Global Regulations

Energy Efficiency

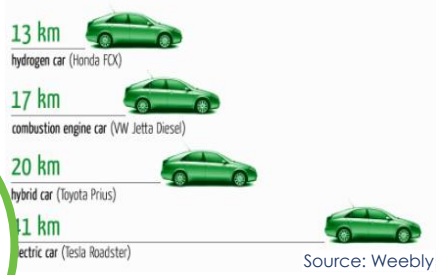
Zero Emission

Multi-Functions



Note: Car refers to average diesel car
Source: BEIS/Defra Greenhouse Gas Conversion Factors 2019

How far can you drive with 10 kilowatthours of energy?
losses from energy production and delivery included (well-to-wheel)



AVERAGE VEHICLE WEIGHT (kg)
VEHICLES SOLD IN EUROPE*
USA 2001-2022

Source: Motor 1

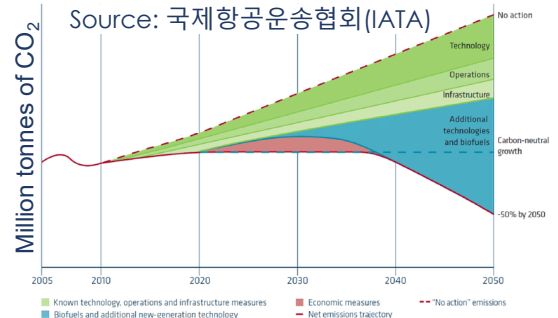


Global Environmental Regulation

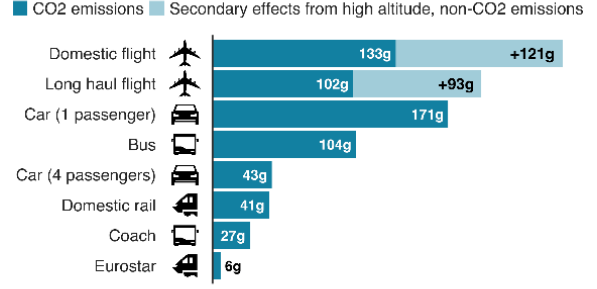
Environmental Social Governance

Regulations and Strategies

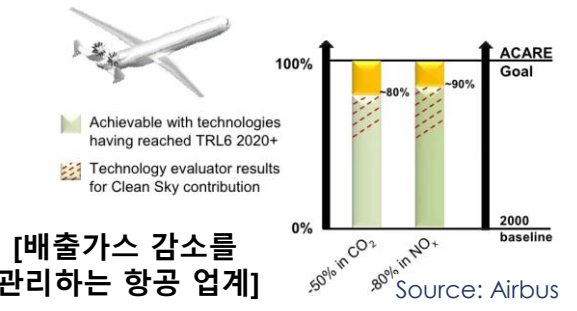
- 탄소중립: 이산화탄소와 같은 탄화가스 규제
- 기후중립: 온실효과에 미치는 모든 가스 규제
- 항공기와 자동차의 매연 저감이 급선무



Emissions per passenger per km travelled



Source: BEIS/Defra Greenhouse Gas Conversion Factors 2019



[배출가스 감소를 관리하는 항공 업계]

Source: Airbus

『2050 탄소중립』 당정협의

일시: 2020. 12. 7. (월) 07:30

장소: 국회 의원회관

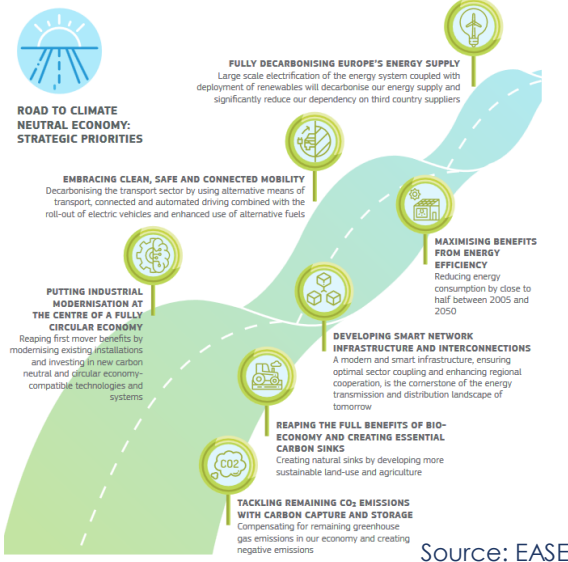
장의위원장

‘2050 탄소중립’ 주요 추진 전략

경제구조 저탄소화	신유망 저탄소산업 생태계 조성	탄소중립 사회로의 공정 전환
<ul style="list-style-type: none"> 화석연료에서 신재생에너지로 에너지 주공급원 전환 철강·석유화학 등 다배출 업종 저탄소 전환 촉진 전기차 충전기 2000만 가구, 수소충전소 2000개 마을·도시 단위 에너지 자립률 제고, 탄소중립도시 조성 	<ul style="list-style-type: none"> 이차전지, 바이오 등 저탄소 산업 육성 친환경·저탄소·에너지 관련 '그린 예비유니콘' 육성 철강·플라스틱 대체 혁신소재 개발 	<ul style="list-style-type: none"> 석탄발전-내연기관차 산업, 대체·유망 분야로 전환 지자체 탄소중립 역량 강화 및 지원 기반 구축 전 국민 대상 환경교육 및 홍보 강화

(자료: 기획재정부)

[탄소중립(Carbon-neutral) 추진 전략]



[기후중립(Climate-neutral) 추진 전략]

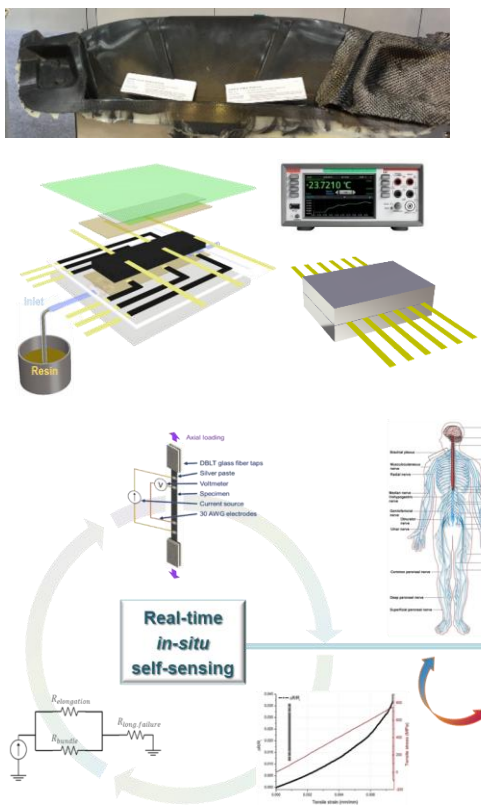
Source: EASE

Introduction

Major Research Interests

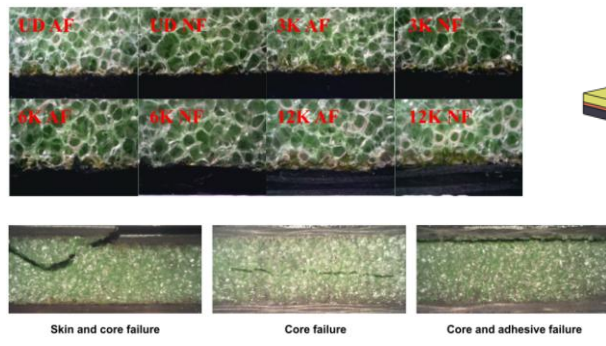
Composite Manufacturing

(Rapid manufacturing, AI optimizing and digitalized real-time manufacturing monitoring)



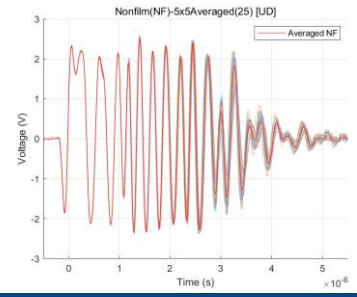
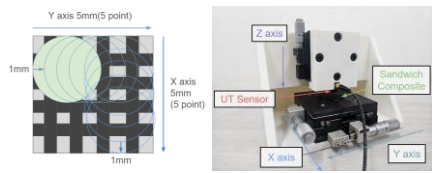
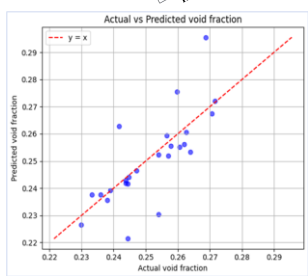
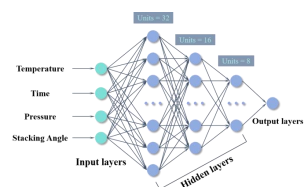
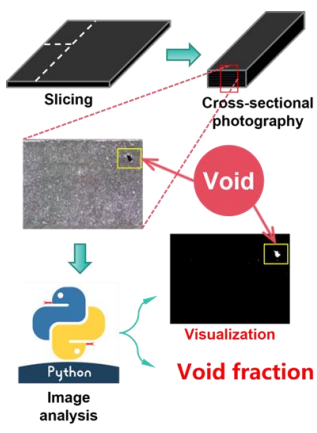
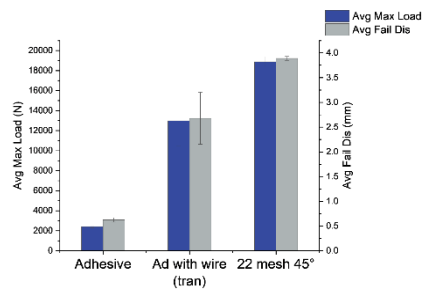
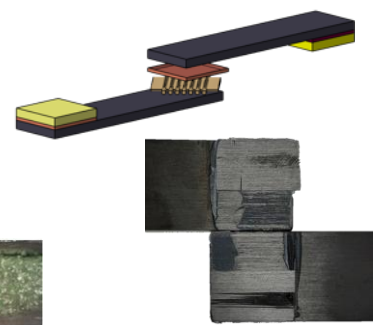
Composite Joining

(Fastener-free joining of CFRTPs and self-assembly of sandwich composites)



Smart Structural Health Self-monitoring

(Structural health monitoring, prognostics and health management and non-destructive evaluation)



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03 FUTURE PLAN

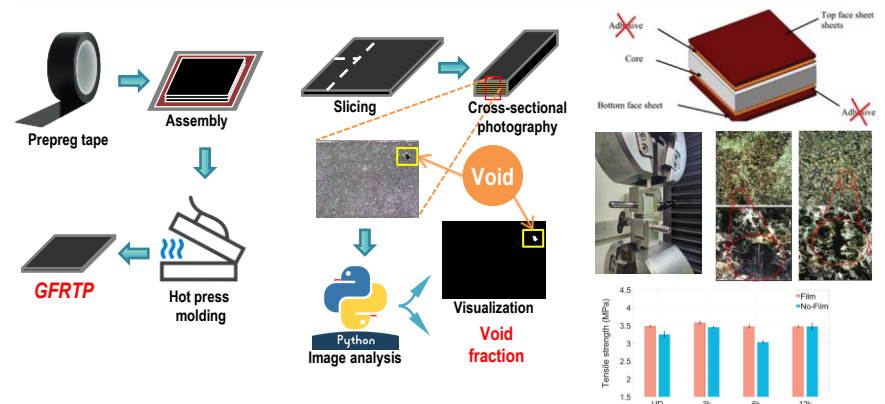
04 SUMMARY

05 APPENDIX

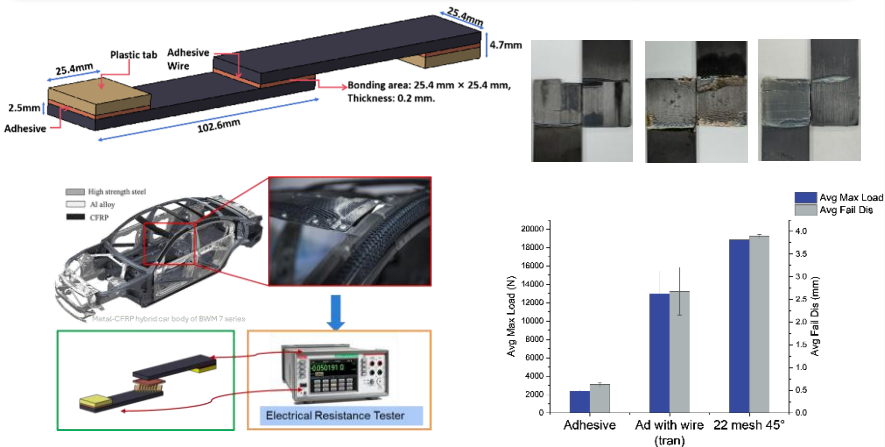
Research Achievements (Overview)

Composite Manufacturing Optimizing & Joining

Manufacturing Optimization

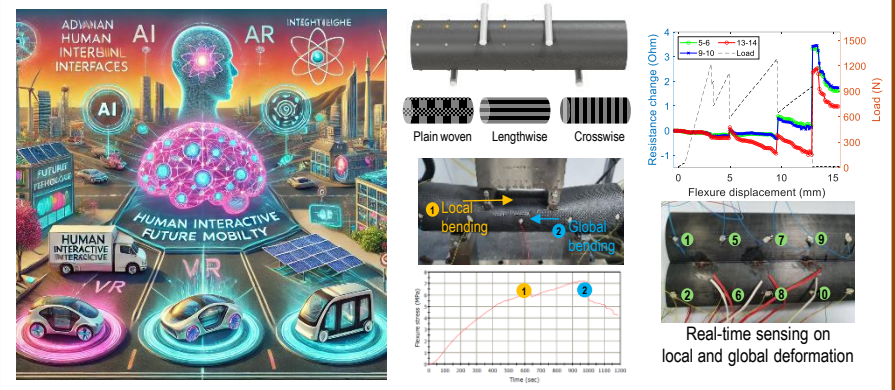


Fastener-free Joining

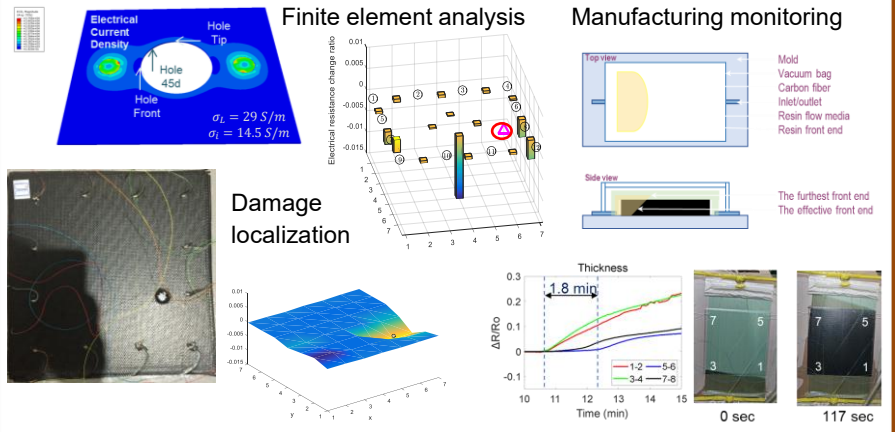


Smart Composites Functionalizing

Functional Composites



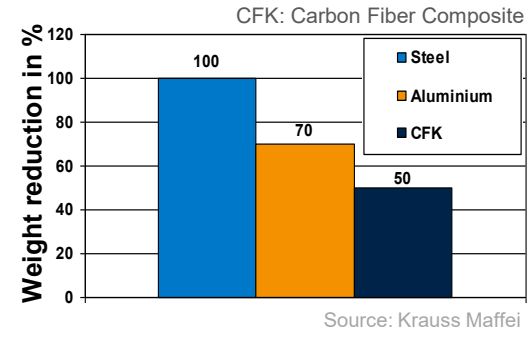
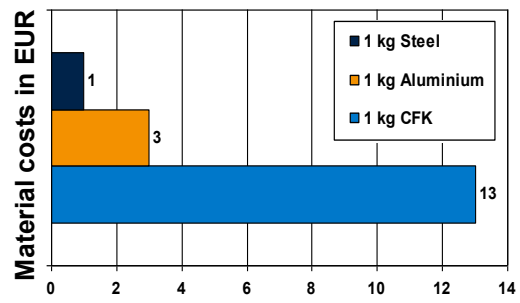
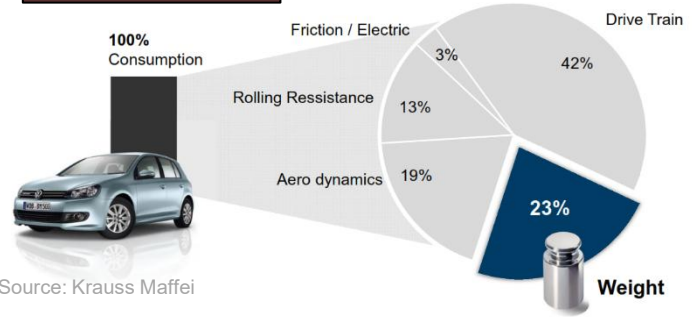
Self-monitoring of CFRPs



Research Achievements (Manufacturing)

Composite Manufacturing

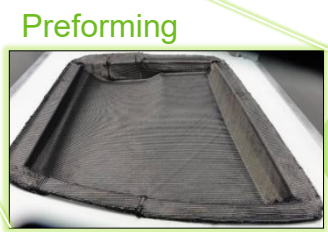
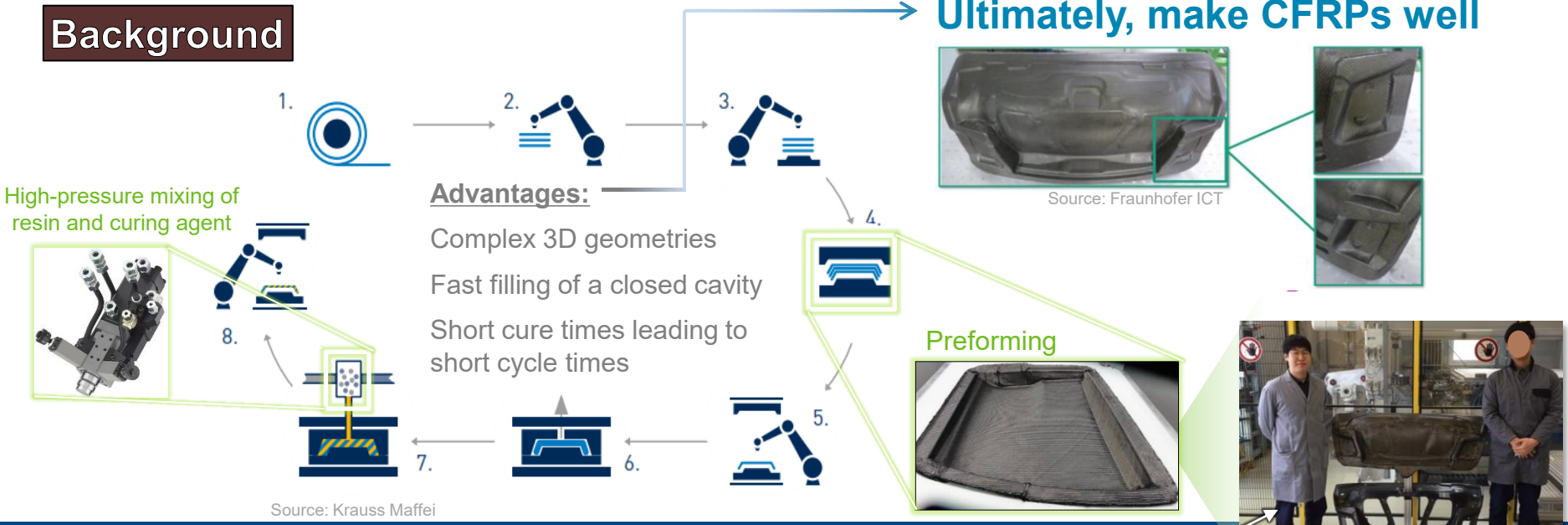
Motivation



Needs for the lightweight material ↑

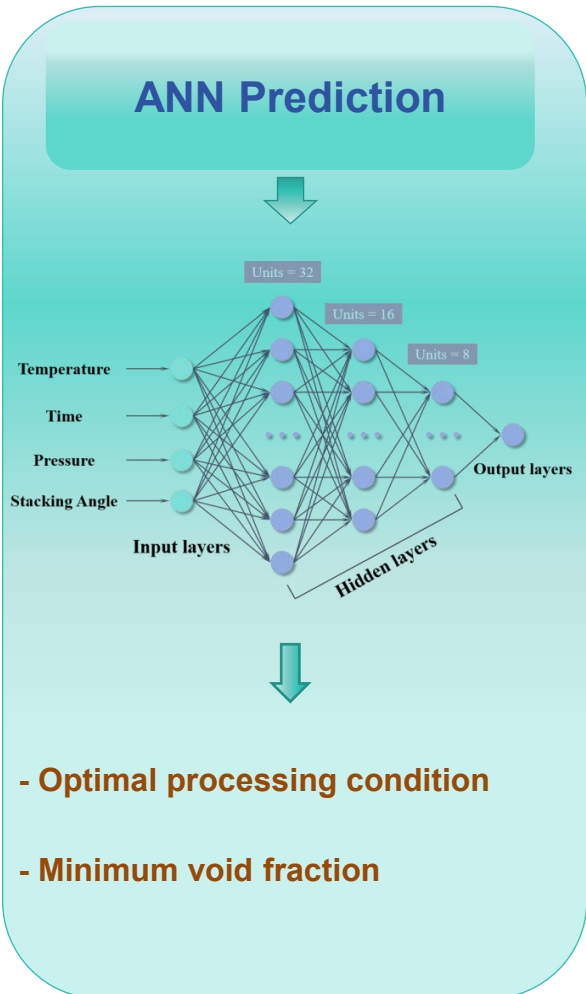
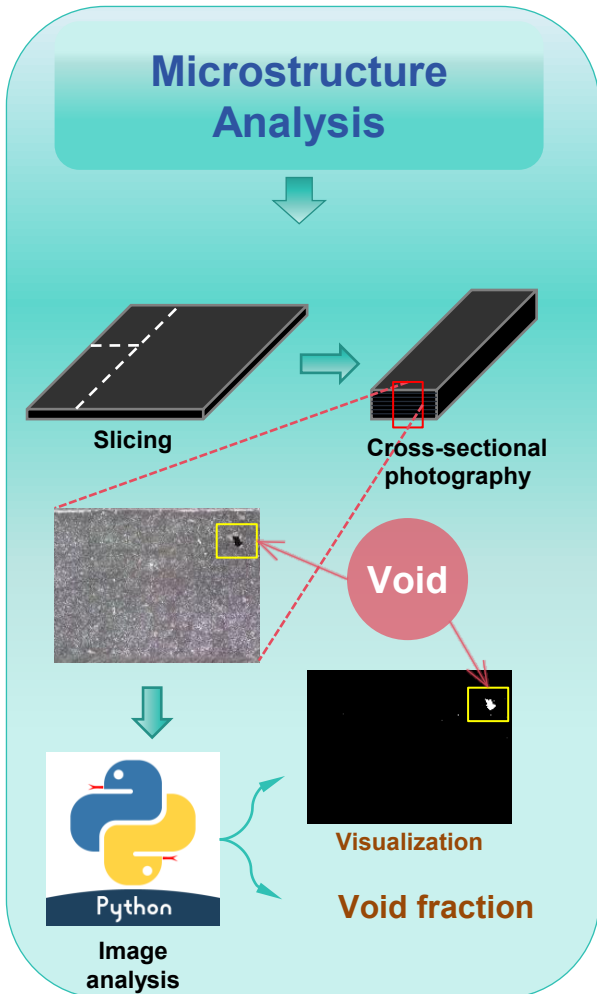
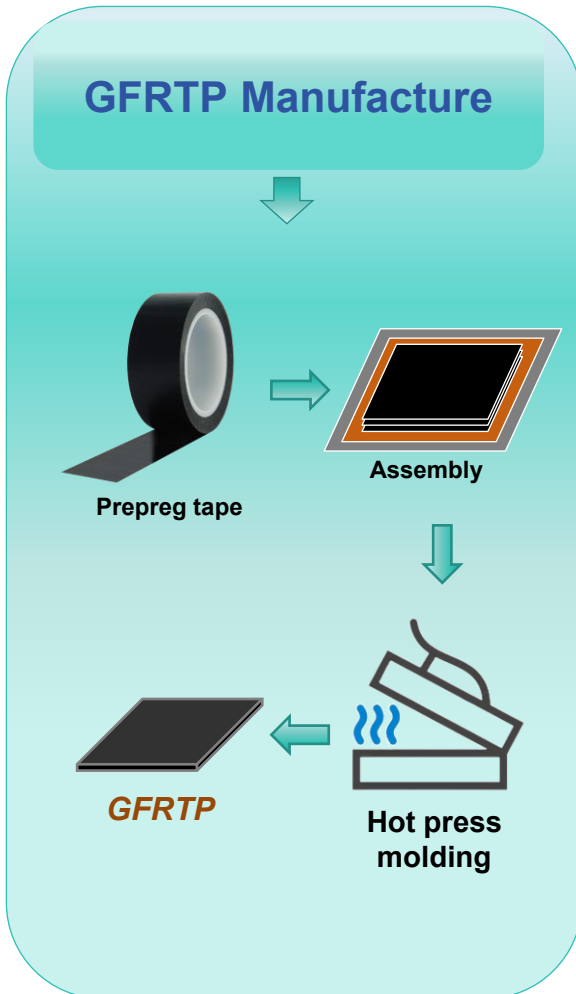
Needs for continuous carbon fiber composite ↑

Background



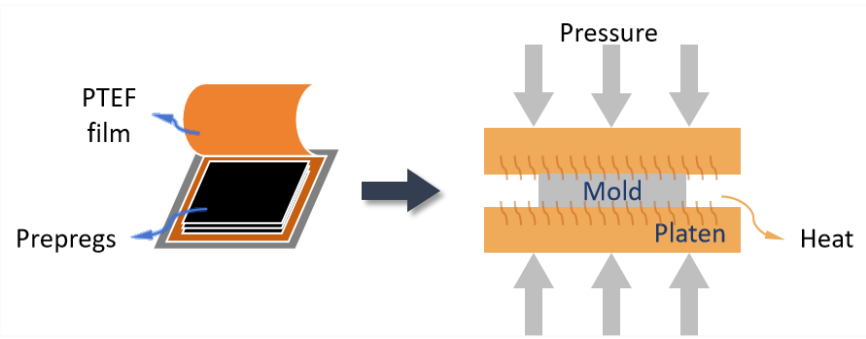
Research Achievements (AI for Optimization)

Prediction of Void Fraction and Optimization of GF RTP Thermoforming Conditions Using ANN



Research Achievements (AI for Optimization)

Prediction of Void Fraction and Optimization of GF RTP Thermoforming Conditions Using ANN



Materials

UD Tape (439 gsm): 70 vol.% E-glass fiber, Celanese
 Resin: PP (Polypropylene)
 Geometry: L 150 mm * W 150 mm * t 0.25 mm

Characterization

Cross-section Analysis

- Cut the test piece with **Rotating wheel**
- **Semi-auto polishing**
- Shoot cross-section by **Digital Optical Microscope**
- Analyze the stacking status and presence of voids
- Analyzing porosity by **Python**

ANN Prediction

- Import Temperature, Pressure, Stacking angle and Heating time as inputs into the ANN model.
- Output minimum void fraction and optimal production conditions.

➤ Cross-sectional Image Analysis

Pressure	void fraction (%)	Left area	Void visualization (L)	Middle area	Void visualization (M)	Right area	Void visualization (R)
30 bar	0.22982						
40 bar	0.21530						
50 bar	0.25396						
60 bar	0.25648						

Research Achievements (AI for Optimization)

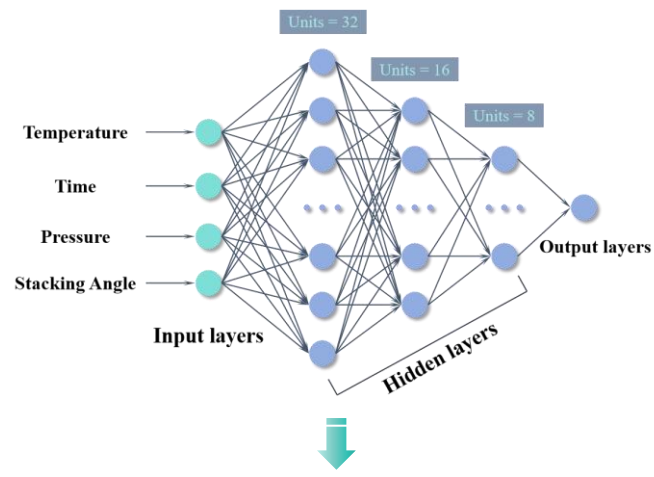
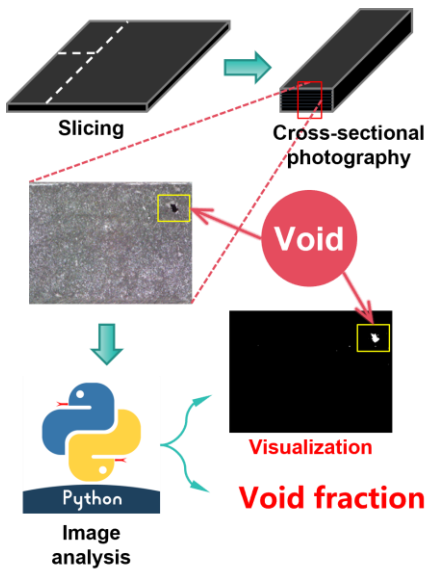
Prediction of Void Fraction and Optimization of GF RTP Thermoforming Conditions Using ANN

Novelties

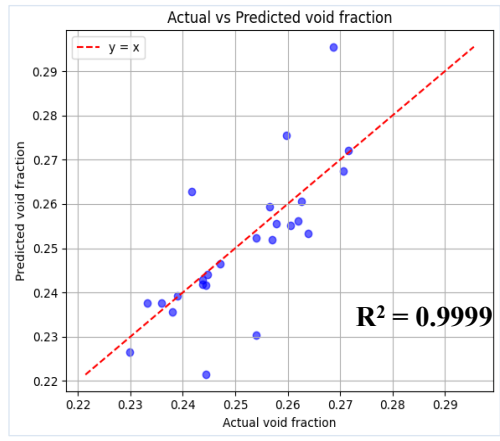
- Emphasis on microstructure analysis.
- ANN-aided optimization.
- Time, cost, and labor minimization
- Process optimization

Improved

- Predicting the optimal process parameters through ANN,
- Improve molding quality;
- Reduce void fraction;
- Enhance product consistency.



- Optimal processing condition
- Minimum void fraction

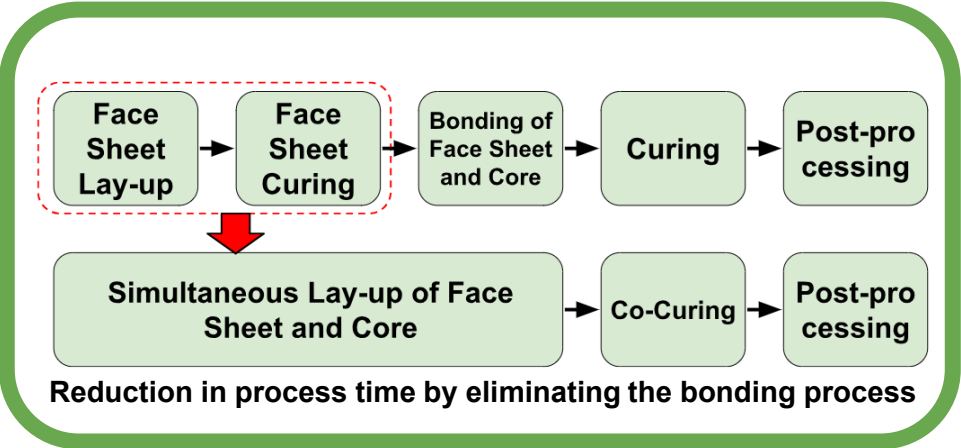


► The proposed ANN models are effective in optimizing GF RTP manufacturing parameters.

Research Achievements (Sandwich Composite)

One-shot Sandwich Composite Co-curing

Co-curing manufacturing



Objectives

1. Process time reduction
2. Manufacturing cost reduction
3. Weight reduction

Adhesive film removal / testing

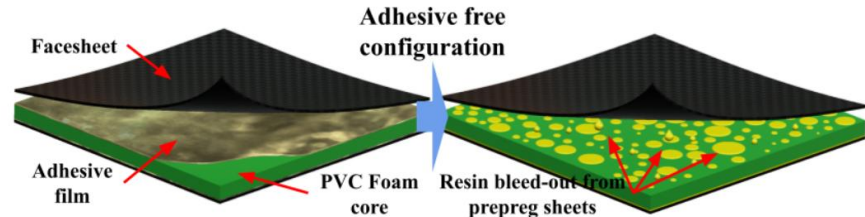
Fiber Type	Film (MPa)	No-Film (MPa)
UD	~3.5	~3.3
3k	~3.5	~3.4
6k	~3.5	~3.2
12k	~3.5	~3.4

Maintaining proper bonding strength even after removing the adhesive film

Research Achievements (Sandwich Composite)

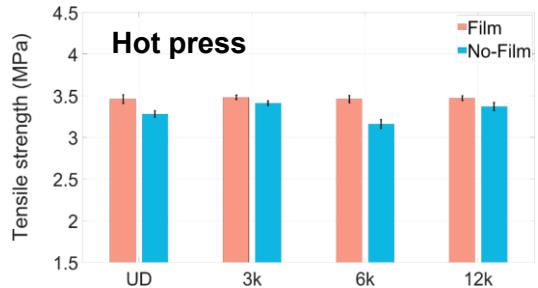
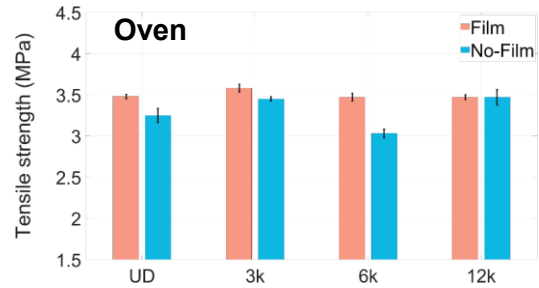
One-shot Sandwich Composite Co-curing

Proposed Method



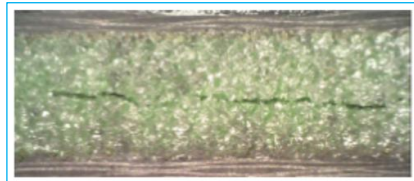
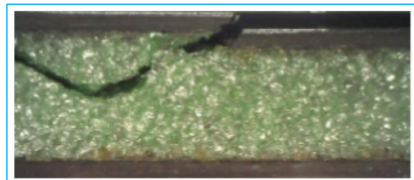
Testing and Results

▶ The proposed co-curing method successfully manufactured sandwich composites.



Specimen	OVEN	
	Film usage	Failure Mode
UD	X	Skin failure and core failure (2) Core failure and adhesive failure (3)
3K	X	Core failure (5)
6K	X	Core failure and adhesive failure (5)
12K	X	Core failure (5)
UD	O	Core failure (5)
3K	O	Core failure (5)
6K	O	Core failure (5)
12K	O	Core failure (5)

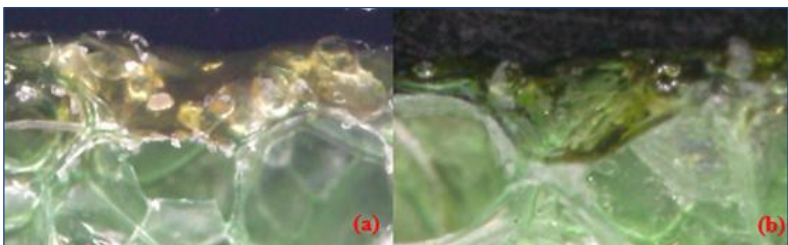
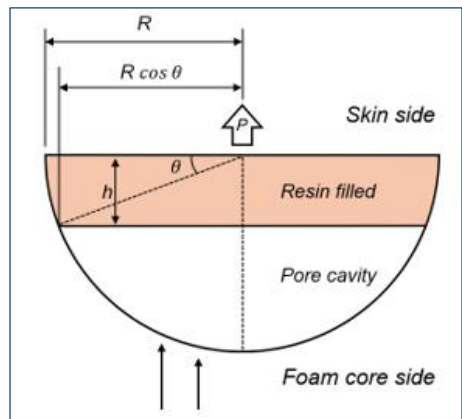
Specimen	Hot - press	
	Film usage	Failure Mode
UD	X	Skin failure and core failure (1) Core failure and adhesive failure (4)
3K	X	Core failure (5)
6K	X	Core failure and adhesive failure (5)
12K	X	Core failure (5)
UD	O	Core failure (5)
3K	O	Core failure (5)
6K	O	Core failure (5)
12K	O	Core failure (5)



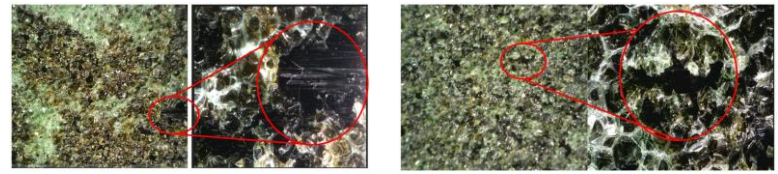
Research Achievements (Sandwich Composite)

One-shot Sandwich Composite Co-curing

Key Technology



Left : Hemispherical-pore infiltration model for pressure-driven resin flow.
 Right : Optical comparison of NF specimens: (a) well-filled, (b) poorly filled.



UD

6K

- ▶ Understanding the resin flow into the core is the key feature to optimize the manufacturing process.
- ▶ The amount of resin impregnation to the foam core is directly related to the joining strength between the skin and the core.

Impregnated depth: $h = R \sin \theta$

Efficient area for bonding: $A_{efficient} = \pi \{R^2 - (R \cos \theta)^2\} = \pi h^2$

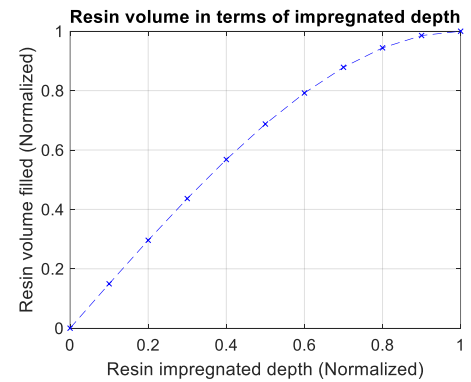
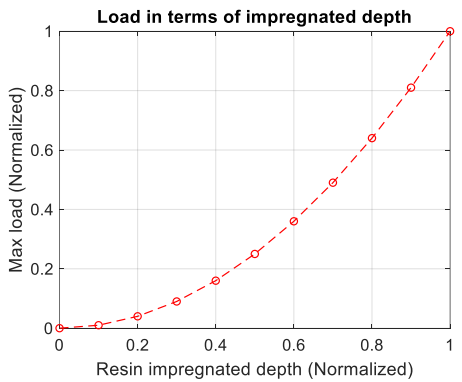
Normal joining strength of the adhesive: $\sigma_{joining} = \frac{P}{A_{efficient}}$

Maximum load: $P = \sigma_{joining} \times A_{efficient}$

Pore cavity volume: $V_{pore} = \frac{\pi}{3} \{(R - h)2 \times (3R - (R - h))\}$

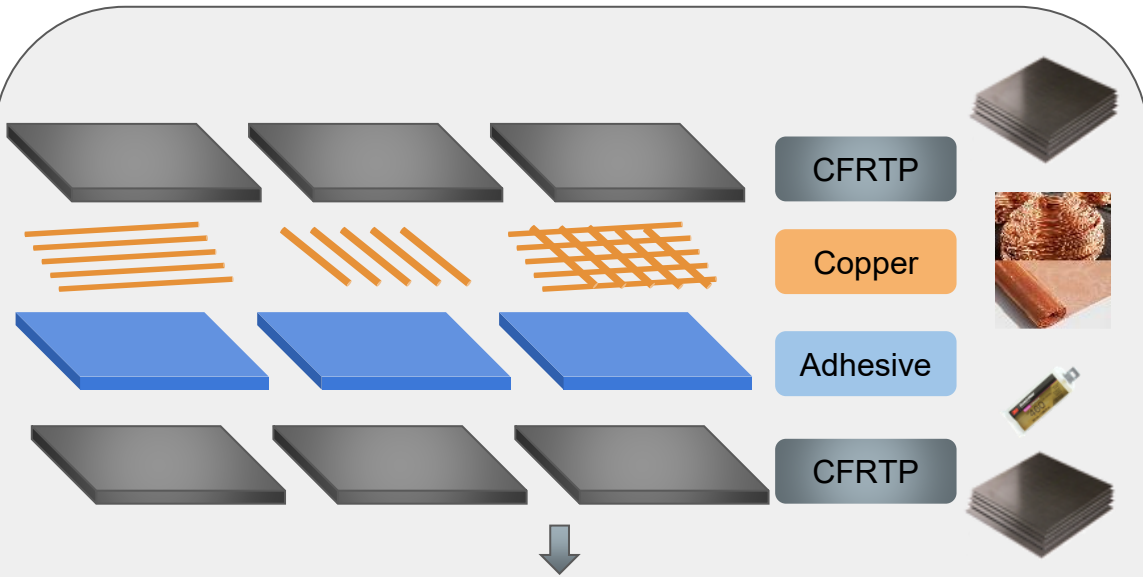
Volume filled: $V_{Filled} = \frac{2\pi}{3} R^3 - V_{pore}$

Viscosity and velocity: $\tau_{yx} = \mu \cdot \frac{du}{dy}$ (Newton's law of viscosity)



Research Achievements (Joining)

Fastener-free Joining



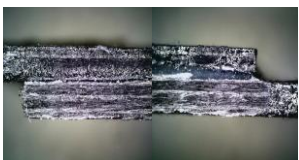
- The CFRTP is polished with sandpaper.
- The polished surface is cleansed with acetone.
- The cleansed CFRTP is cured in high-temperature vacuum environment.

Research Features:

- High strength
- High conductivity
- Lightweight

Key Improvements:

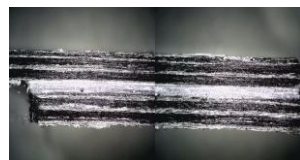
- Copper mesh reinforces joining
- Lightweightness with functional and structural components



Adhesive



Adhesive & oven



Adhesive wire & oven (long)

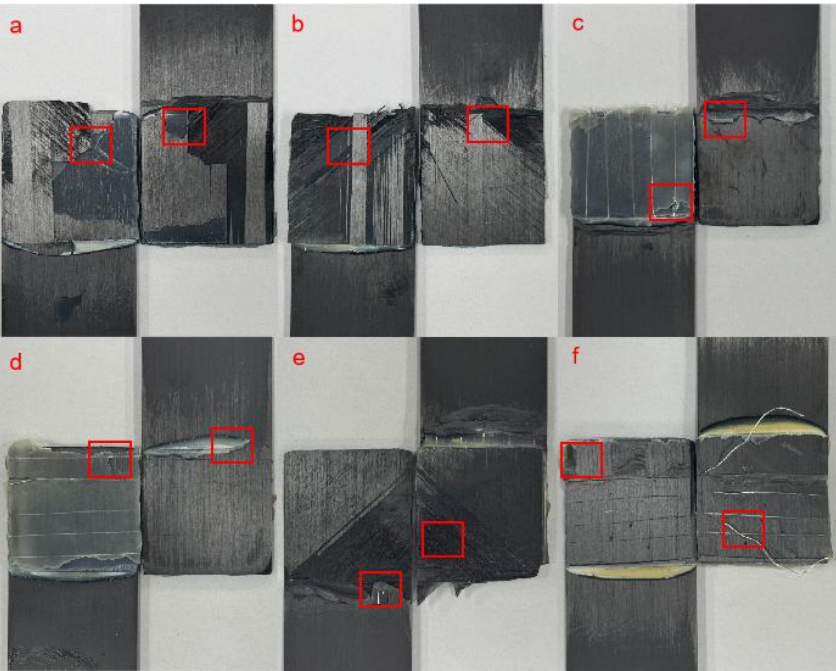
► **CFRTPs were joined with copper wire (or mesh) and adhesive.**

► **The copper material reinforced the joining strengths between CFRTPs.**

Research Achievements (Joining)

Fastener-free Joining

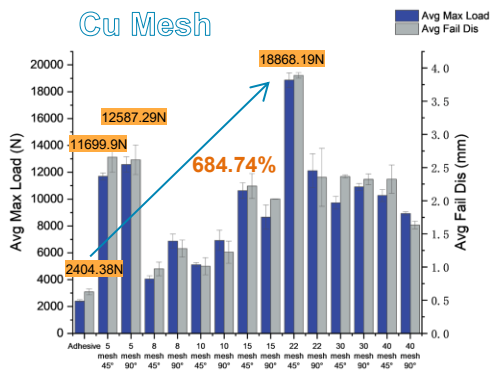
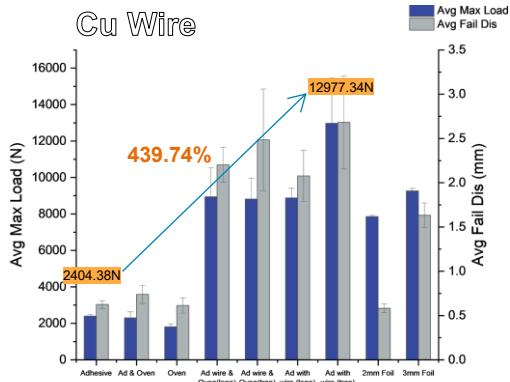
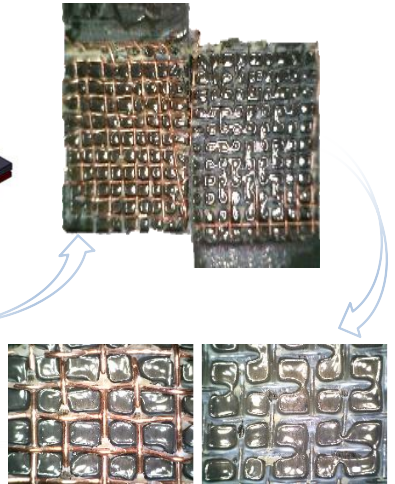
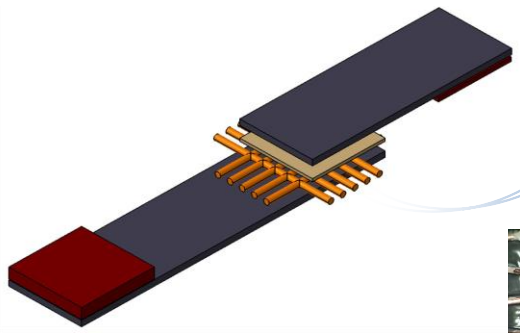
Cu Wire Joining



a. Cohesive failure b. Substrate failure c. Adhesive failure
 d. Mixed failure e. Delamination f. Mixed failure

- ▶ The proposed copper aided joining showed superior joining strengths.
- ▶ The copper materials sustained the structural loads at the joining interface.

Cu Mesh Joining



- ▶ The proposed joining method represented 684.74% higher joining strength.

Research Achievements (Non-destructive Evaluation)

Ultrasonic Non-destructive Testing

Durability of C/GFRPs

UT (MHz)
Active system (Pulse)
Monitoring failures

AE (kHz)
Passive system (Event)
Requires external loading

- FRP degradation mechanism diagram in sea water

Evaluate / Model / Estimate the Structural Endurance

- Real time FFT signal processing using Simulink
- Modeling for remaining useful lifetime using CNN
- Optimized image processing for NDE systems

Scan and Analysis

Acousto-Ultrasonic scan

- Synergistic combination of AE and UT in non-destructive evaluation

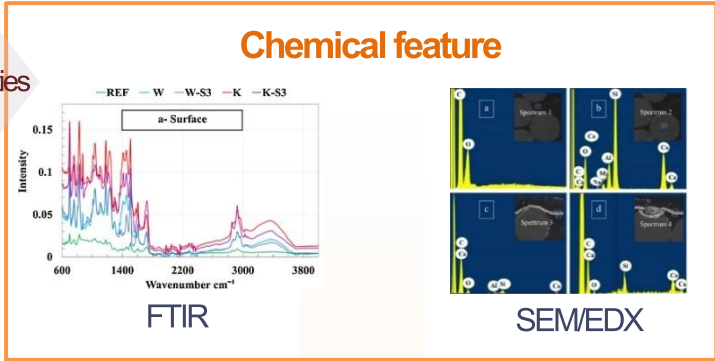
Analysis

Research Achievements (Non-destructive Evaluation)

Ultrasonic Non-destructive Testing

Original C matrix (by TDS)

$$C = \begin{bmatrix} C_{11} & C_{12} & C_{13} & 0 & 0 & 0 \\ C_{12} & C_{22} & C_{23} & 0 & 0 & 0 \\ C_{13} & C_{23} & C_{33} & 0 & 0 & 0 \\ 0 & 0 & 0 & C_{44} & 0 & 0 \\ 0 & 0 & 0 & 0 & C_{55} & 0 \\ 0 & 0 & 0 & 0 & 0 & C_{66} \end{bmatrix}$$



Properties

Compare with

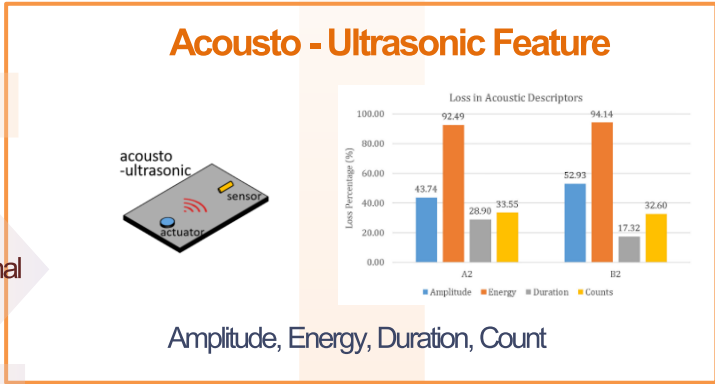
Compare with

Inverse C matrix calculation (by wave velocity)

Component	Velocity type	Direction
E_1	V_{S0}^{grp}	0°
E_2	V_{S0}^{grp}	90°
E_3	V_L^{bulk}	z
G_{12}	V_{ASH1}^{grp}	in-plane 45°
G_{13}	$V_{45,1-3}$ (SSH1/bulk)	$0^\circ-z$
G_{23}	$V_{45,2-3}$ (SSH1/bulk)	$90^\circ-z$

In plane (for E_1, E_2)
Out of plan (for E_3)

*Angular Incidence (mode selection)



Data

Signal

Degraded C matrix (by aging)

Molecular mechanics

Degrading coefficients

Finite Element Method with updated C matrices

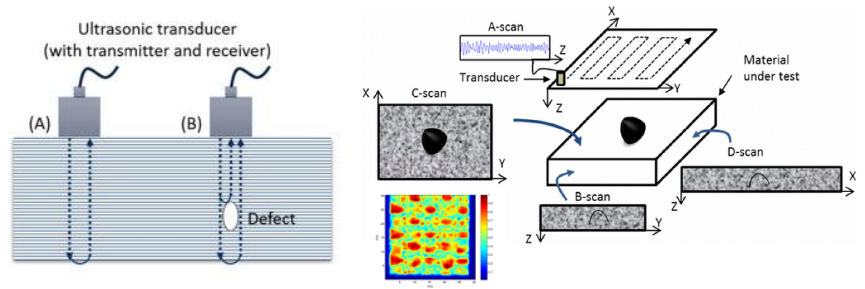
Advanced Remaining Useful Lifetime

Research Achievements (Non-destructive Evaluation)

Ultrasonic Non-destructive Testing

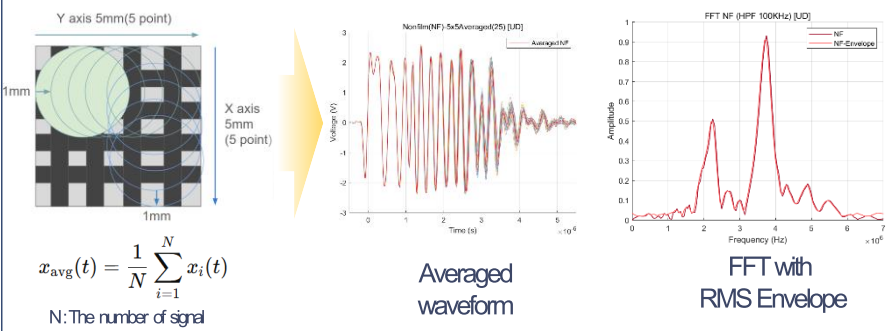
Ultrasonic testing (Sole probe)

- Identifies defects by analyzing reflected echo signals
- Low resolution and easy to analyze
- Good for crack detection and imaging



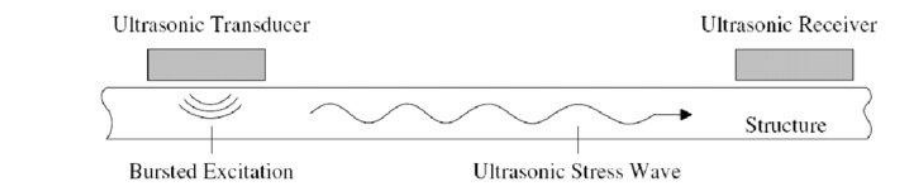
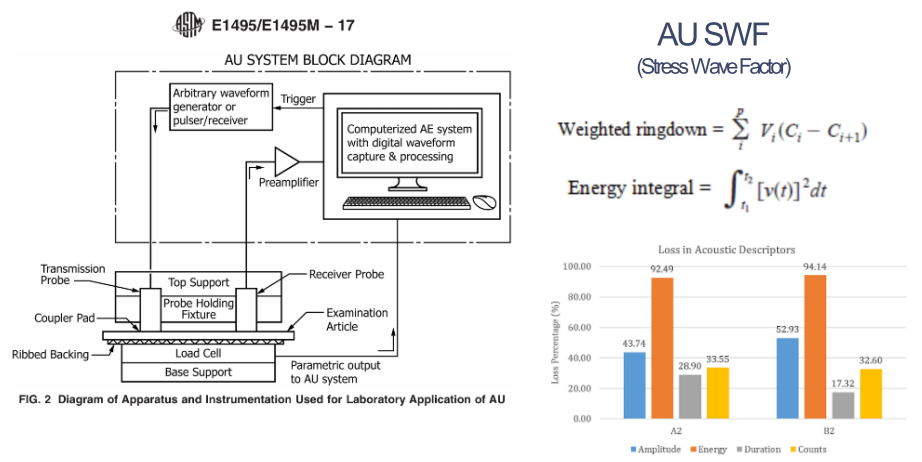
Normalizing A-scan signal

- 5x5(25) point array inspection



Acousto-Ultrasonic testing (Paired probes)

- Detects internal micro-damage through stress wave energy
- High resolution, Hard to analyze
- Optimized for evaluating mechanical properties of materials



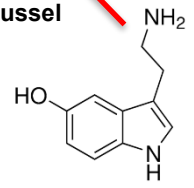
Research Achievements (Skin-Foamcore Self-assembly)

Polydopamine based Skin-Foamcore Joining

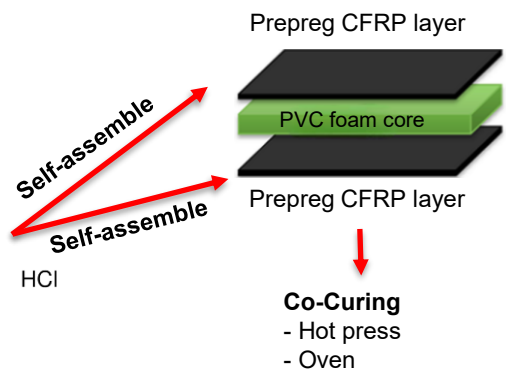
Introduction



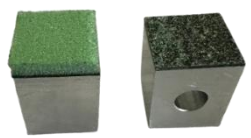
Discovered in mussel



Dopamine hydrochloride



Results



Delamination

Reason

- The epoxy dissolves in acid
- The epoxy absorbed moisture, reducing adhesion
- The immersed CFRP was dried at 60°C, and partial curing proceeded, resulting in lower adhesion

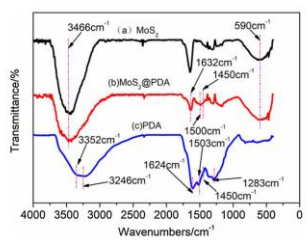
Plan

- Dry CFRP completely at room temperature
- Experiment with various conditions

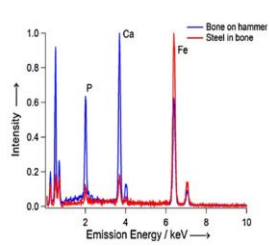
Analysis Method

Chemical reaction

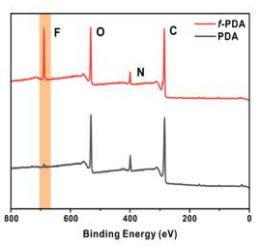
FTIR



SEM-EDX



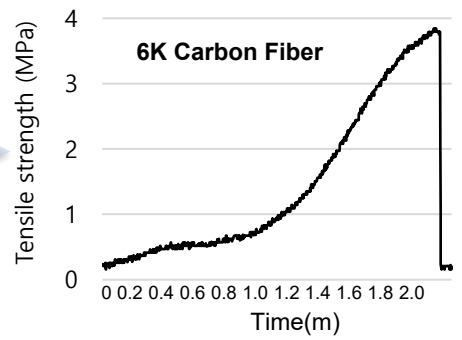
XPS



Tensile strength



Tensile testing(Z-dir)



Structural Health Monitoring

Structural Duty and Safety

- Structural Health Monitoring
 - Safety, cost, and operating efficiency
 - Sensing element, method, and cost

Aloha airlines aircraft failure (1988)

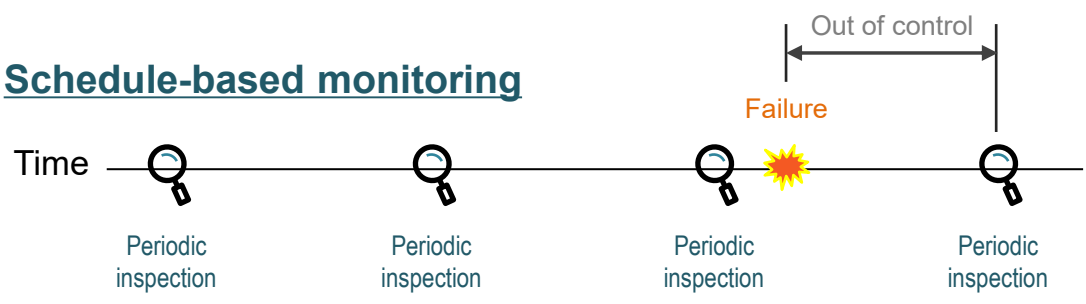


35W bridge collapse (2007)

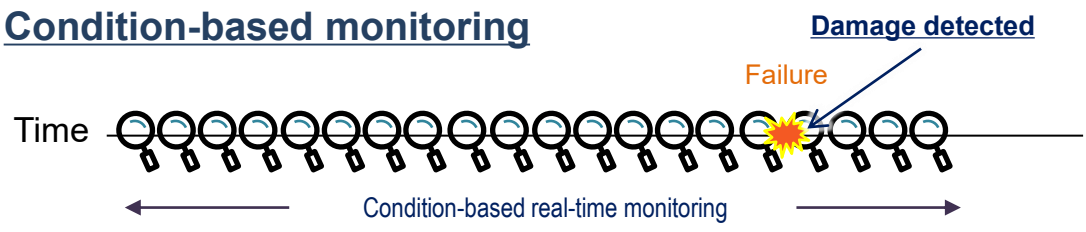


- Condition-based Monitoring
 - Real-time, low cost, and easy installment
 - Inspection under in-service

Schedule-based monitoring

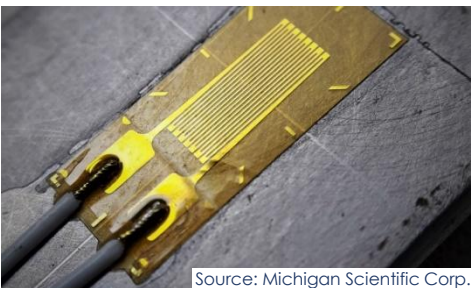


Condition-based monitoring



Structural Health Monitoring

Strain gauge

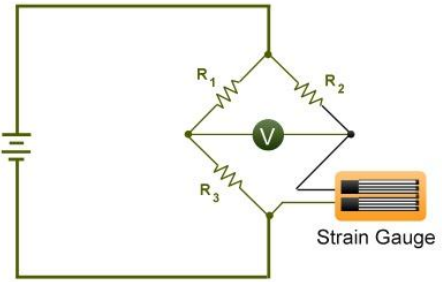


Source: Michigan Scientific Corp.



Source: Binsfeld

Quarter-bridge Strain Gauge Circuit

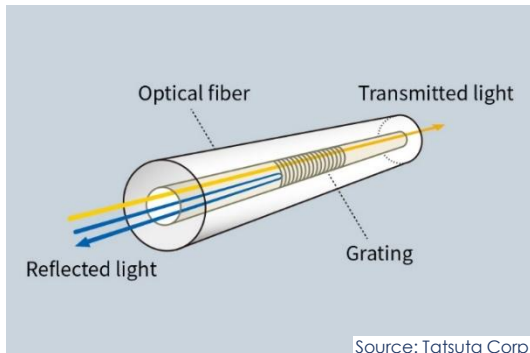


© Chipkin Automation Systems Inc.

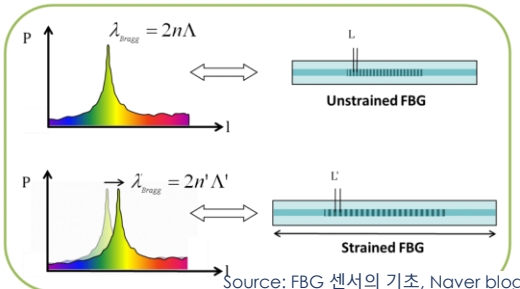
Fiber Bragg grating



Source: Fiber Transceiver Solution Corp.

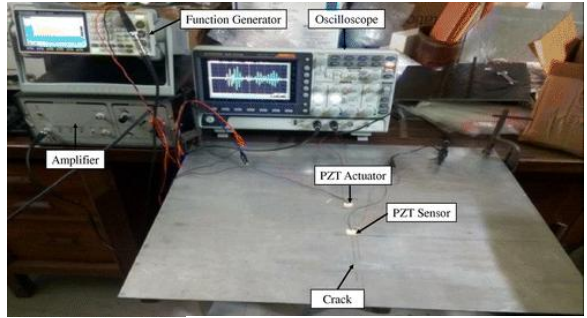


Source: Tatsuta Corp.

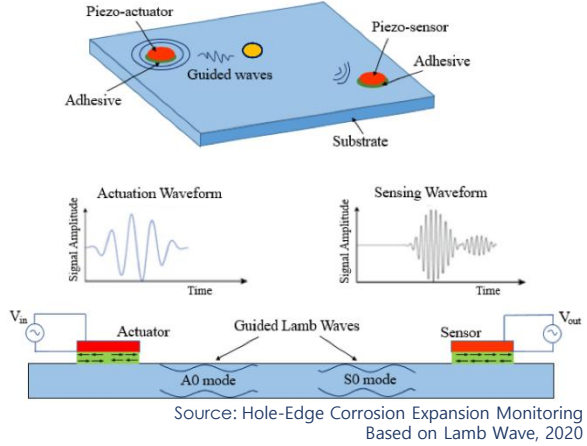


Source: FBG 센서의 기초, Naver blog

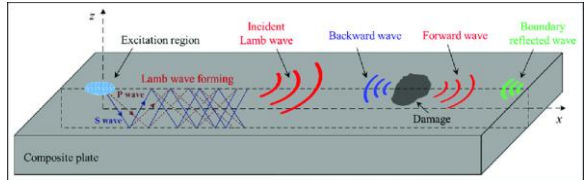
Lamb wave



Source: Data-Driven Lamb-Wave-Based Approach to Detect Multiple Structural Damages, 2021



Source: Hole-Edge Corrosion Expansion Monitoring Based on Lamb Wave, 2020



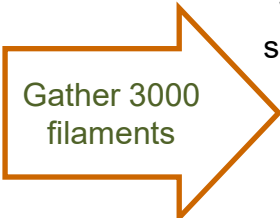
Source: Barely visible impact damage imaging using non-contact air-coupled transducer/laser Doppler vibrometer system, 2016

Research Achievements (Smart composites for SHM)

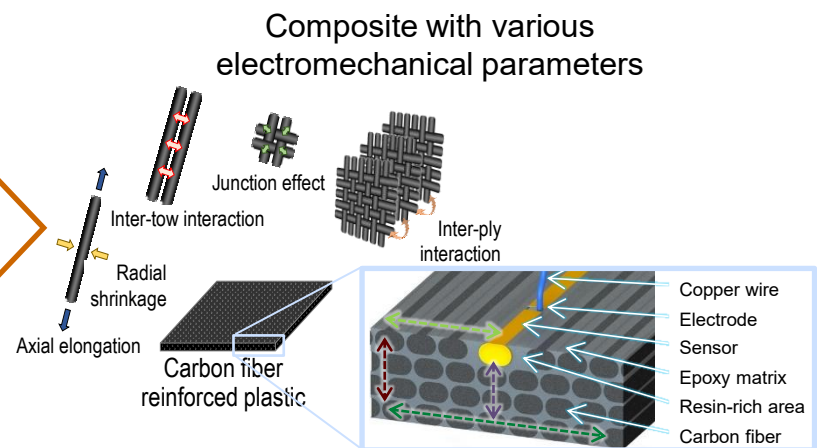
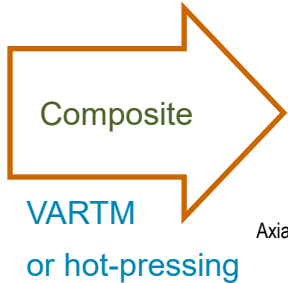
Electromechanical Analysis of CFRPs

Overview

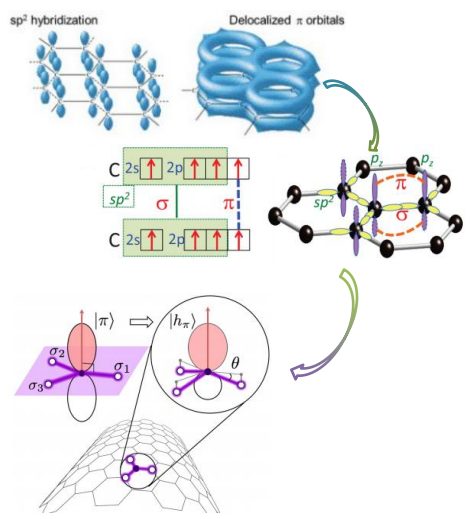
$\varnothing 8 \mu m$ Carbon fiber monofilament



3K carbon fiber single tow bundle



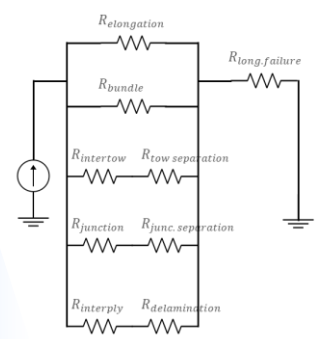
Change in π orbitals and electrical networks



Multi-scale treatment and electrically equivalent circuit modeling

Schematic	R-equiv. factor	Schematics	Trends
	$R_{elongation}$		
	R_{bundle}		
	$R_{failure}$		
	$R_{inter-tow}$		
	$R_{junction}$		

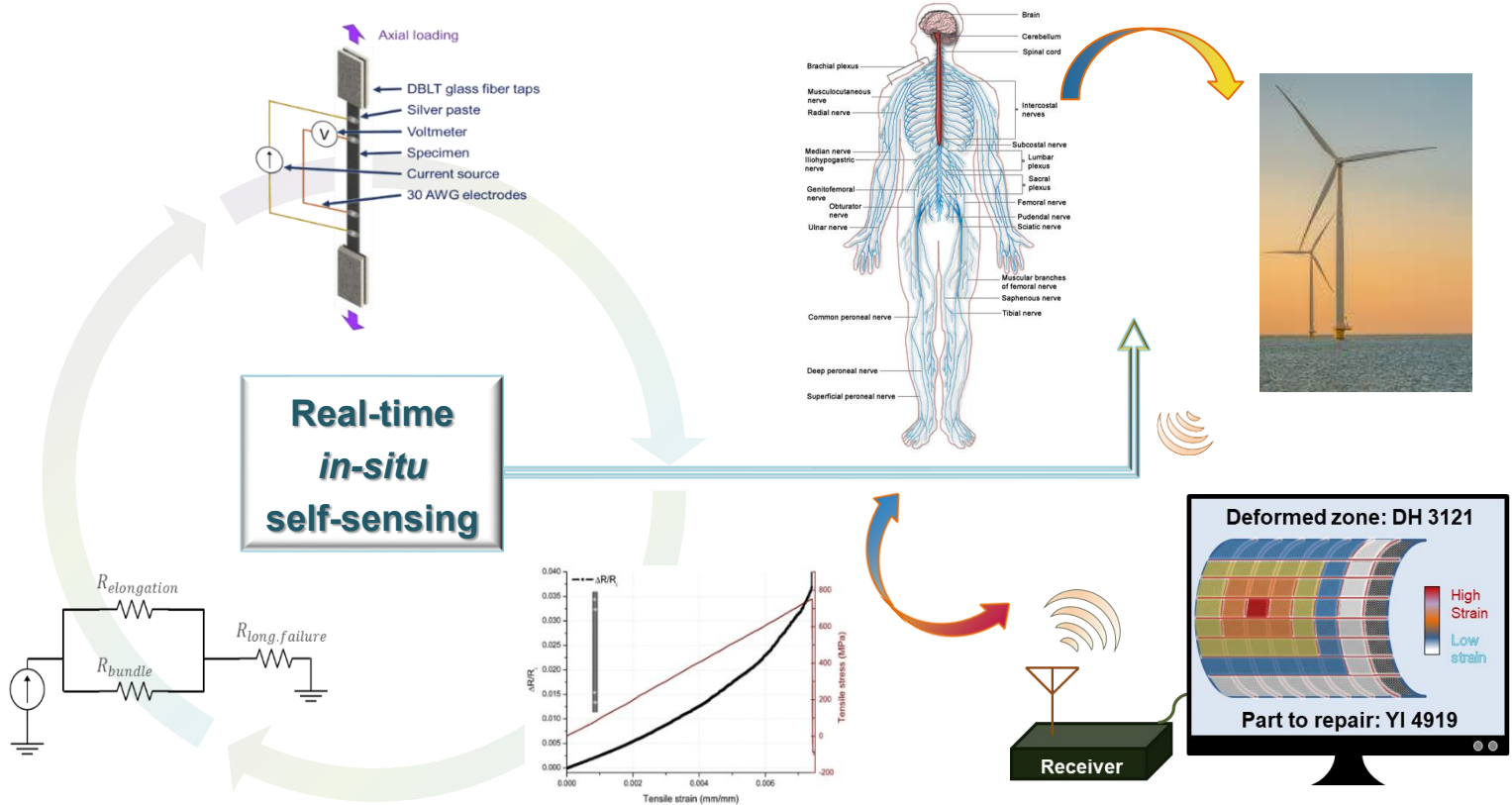
Electrically equivalent circuit modeling



- Elastic deformation
- ↔ Electrical resistance

Research Achievements (Smart composites for SHM)

Research Objectives



- Electromechanical behavior of CFRP enables self-sensing.
- Smart CFRP is reliable with real-time self-sensing system.

Monitoring CFRP Manufacturing

Motivation

► CFRP manufacturing is high-cost, labor-intensive, and time-consuming.

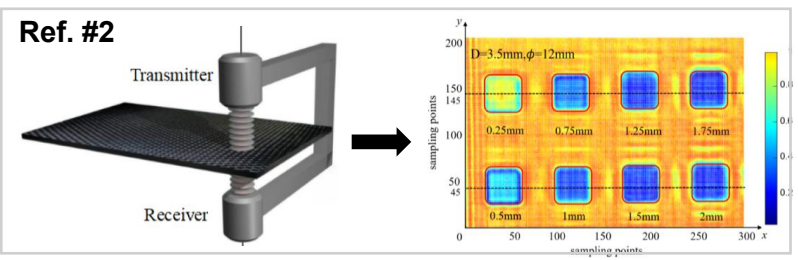
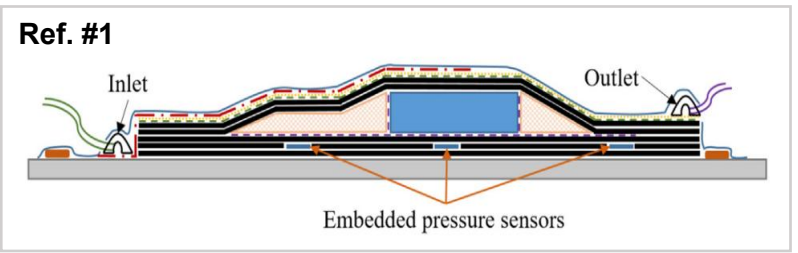
Limitations from the Literatures

1. Embedded Sensor

- Expensive, Possible of damage,
- Cause of stress concentration.

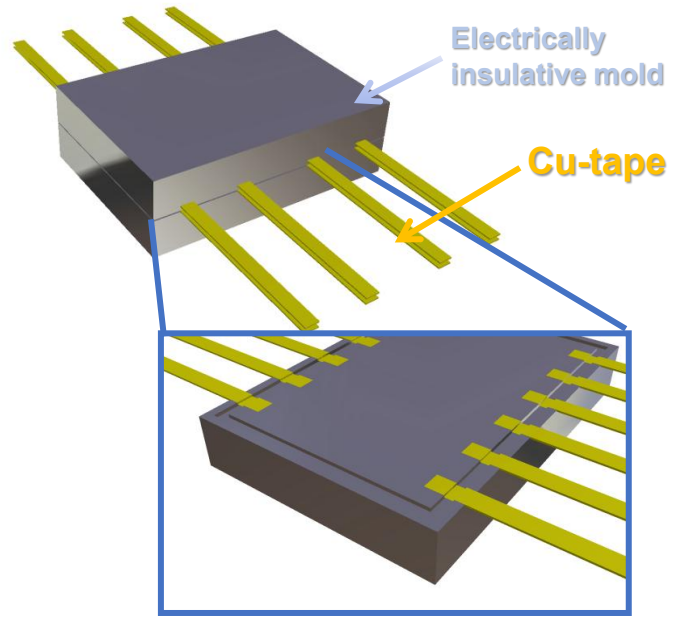
2. *A posteriori* inspection system

- Absence of real-time monitoring.



Novelty and Originality

1. Low cost
2. Simple installation
3. Real-time



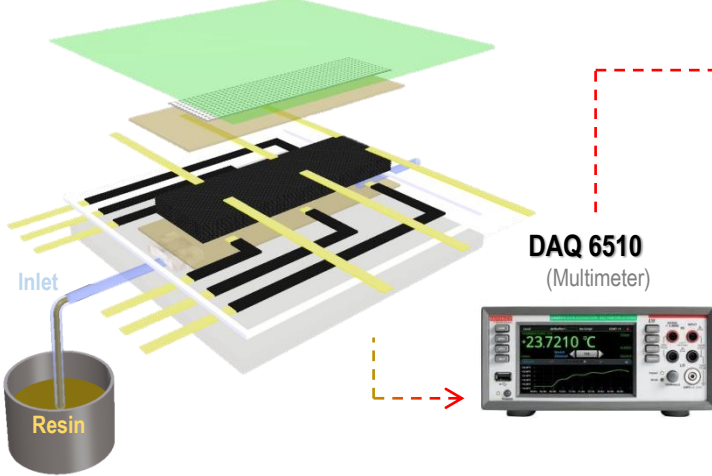
#1. A. Dimassi, *Materials Today : Proceedings*, (2021),140-8
#2. Chenchen Zhang et al., *Ultrasonics*, (2023), 106884

Research Achievements (Manufacturing monitoring)

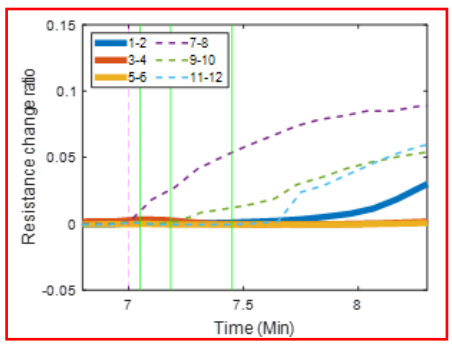
Monitoring CFRP Manufacturing

Overview

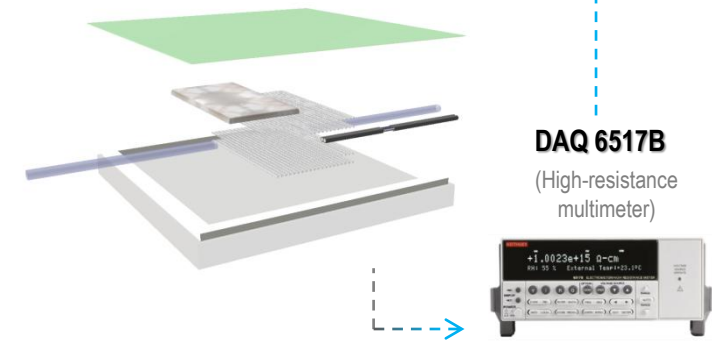
① Electrode analysis



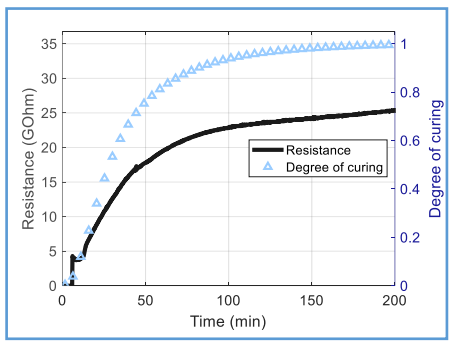
② Resin flow monitoring



③ Resin curing



④ Resin curing monitoring



① Electrode analysis

- Electrode material:
Silver paste vs. Cu-tape
- Pressure and contact resistance

② Resin flow monitoring

- Electrical resistance changes
- Resin front detection
- Degree of impregnation (thickness)

③ Resin curing

- In-mold resin reservoir system
- Customized electrical probe

④ Resin curing monitoring

- Differential scanning calorimetry (DSC)
- Polymer crosslinking and electrical conductivity changes

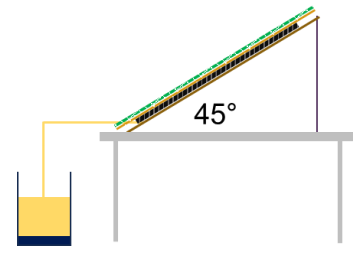
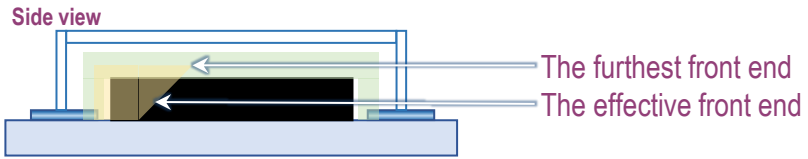
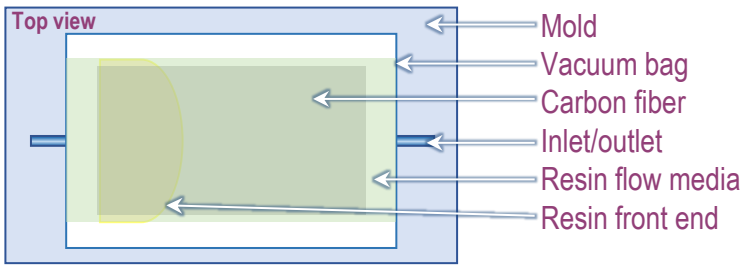
Ultimately,

Real-time composite-manufacturing monitoring mold

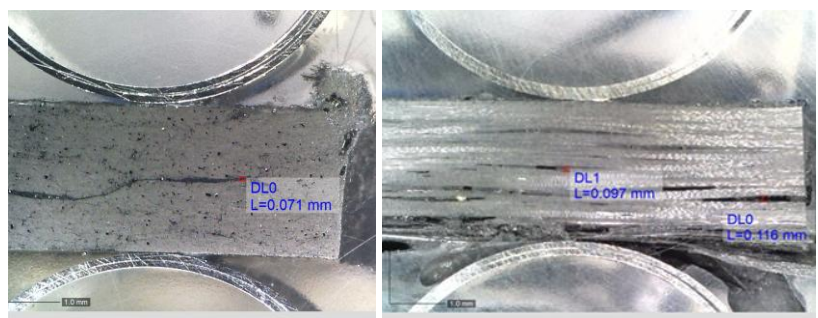
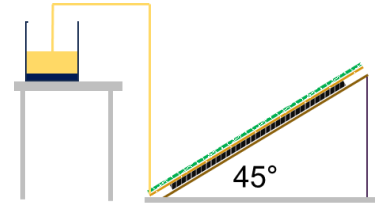
Detect resin impregnation and curing using electrical resistance

Monitoring CFRP Manufacturing

Experimental Analyses

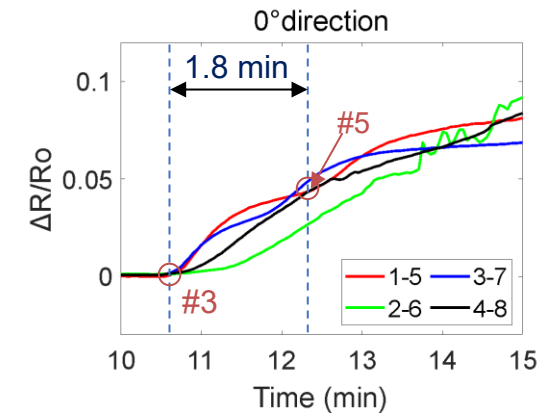
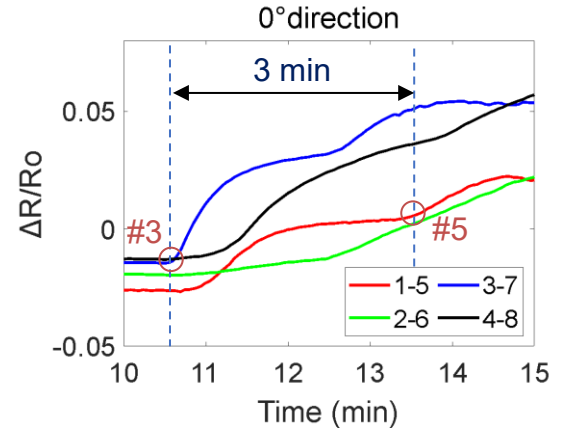


$$\frac{v^2}{2} + gz + \frac{p}{\rho} = \text{constant}$$



Poorly impregnated

Well impregnated



► CFRP manufacturing considering Bernoulli's equation and void contents was investigated to improve the composite quality: Time and void.

Thermal and Electromagnetic Effectiveness

Electric Vehicle Communication Controller (EVCC)

EVCC is a device that communicates with charging stations on rechargeable electric vehicles.

➤ Key Functions of an EVCC

- **Communication Management**

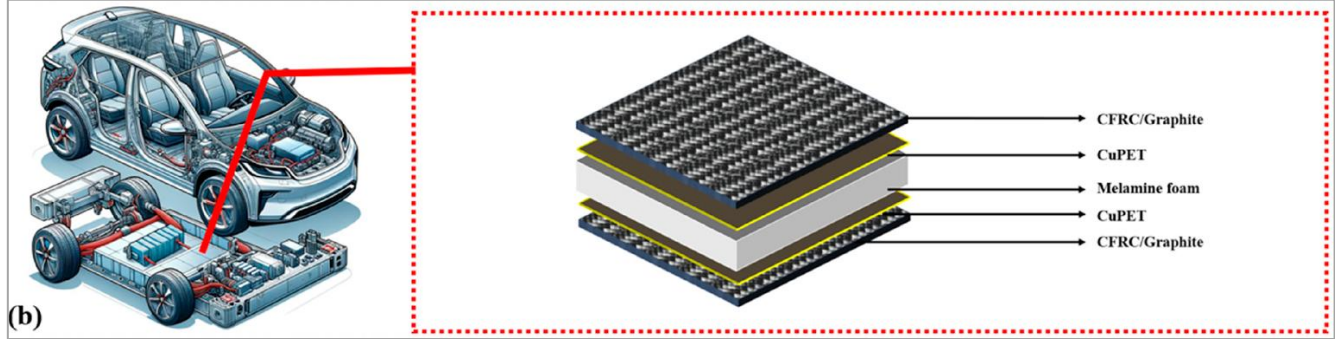
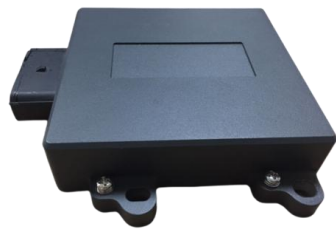
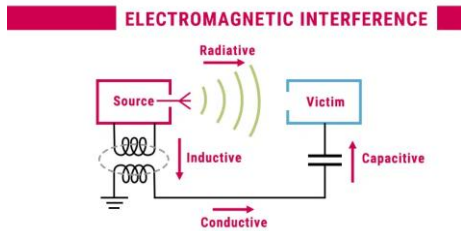
- The EVCC acts as a bridge, enabling seamless communication between the vehicle and the charging station.

- **Power Control**

- It manages the flow of electricity, controlling voltage and current levels to ensure optimal and safe charging.

- **Charging Management**

- The EVCC handles the overall charging process, including starting and stopping charging, and monitoring charging parameters.

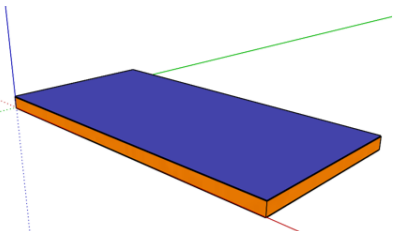


Hu, Shi, et al. (2024). Electromagnetic interference (EMI) shielding and thermal management of sandwich-structured carbon fiber-reinforced composite (CFRC) for electric vehicle battery casings. *Polymers*, 16(16), 2291.

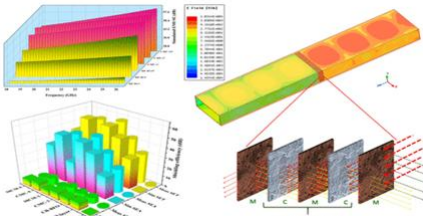
Thermal and Electromagnetic Effectiveness

[As-is]

Electromagnetic shielding performance verification



- Electromagnetic shielding performance according to the difference in thickness of the conductive layer

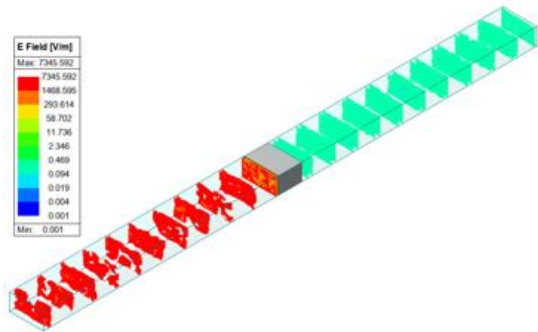


- Application of MG model to electromagnetic wave analysis of composite materials
- Shielding/absorption analysis

[Break-through]

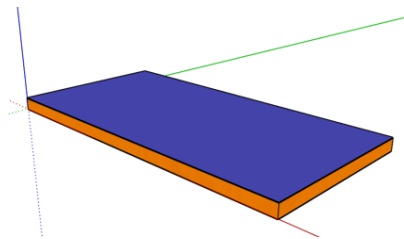
$$\begin{bmatrix} N_x \\ N_y \\ N_{xy} \\ M_x \\ M_y \\ M_{xy} \end{bmatrix} = \begin{bmatrix} A_{11} & A_{12} & A_{16} & B_{11} & B_{12} & B_{16} \\ A_{12} & A_{22} & A_{26} & B_{12} & B_{22} & B_{26} \\ A_{16} & A_{26} & A_{66} & B_{16} & B_{26} & B_{66} \\ B_{11} & B_{12} & B_{16} & D_{11} & D_{12} & D_{16} \\ B_{12} & B_{22} & B_{26} & D_{12} & D_{22} & D_{26} \\ B_{16} & B_{26} & B_{66} & D_{16} & D_{26} & D_{66} \end{bmatrix} \begin{bmatrix} \epsilon_x \\ \epsilon_y \\ \gamma_{xy} \\ \kappa_x \\ \kappa_y \\ \kappa_{xy} \end{bmatrix}$$

- Analysis of electromagnetic shielding through extension of CLPT theory
- Developing an automated electromagnetic shielding analysis tool



- Electromagnetic shielding analysis using Ansys HFSS, Twin Builder / Circuit Simulator, Slwave, CST studio

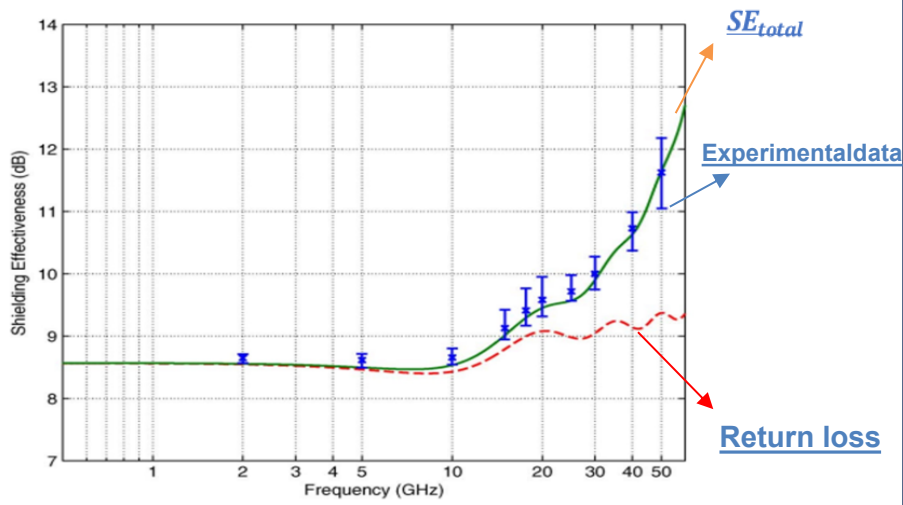
[To-be]



- Conductive layer with appropriate thickness
- Conductive layer material with high electromagnetic shielding performance efficiency
- Equivalent model for thermal and electromagnetic analysis with finite element analysis

Thermal and Electromagnetic Effectiveness

Current method



- **Results of measuring/simulating actual S-parameters**

Theory

$$SE_{total}(f) = R(f) + A(f)$$

- **Return loss**

$$R(f) = 20 \log_{10} |\Gamma(f)|, \quad \Gamma(f) = \frac{Z_{mat}(f) - Z_0}{Z_{mat}(f) + Z_0}$$

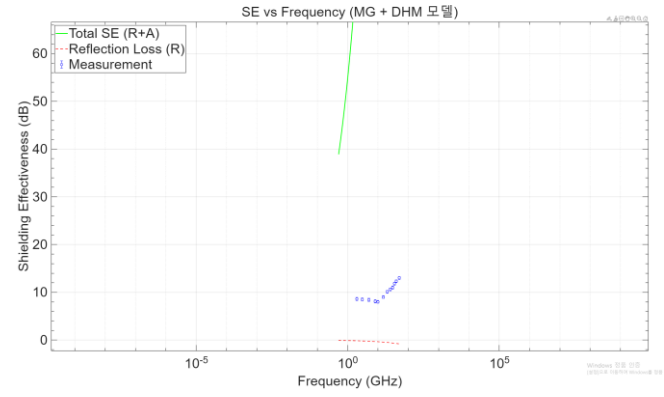
$$Z_{mat}(f) = \sqrt{\frac{\mu_0}{\epsilon_0 \epsilon_{eff}(f)}}, \quad Z_0 \approx 377 \Omega$$
- **Absorption loss**

$$A(f) = 8.686 \alpha(f) d, \quad \alpha(f) = \Re\{\gamma(f)\},$$

$$\gamma(f) = j\omega \sqrt{\mu_0 \epsilon_0 \epsilon_{eff}(f)}$$

MG + DHM model

Our achievement



$$\epsilon_{eff} = \epsilon_m \cdot \left(\frac{\epsilon_f + 2\epsilon_m + 2f(\epsilon_f - \epsilon_m)}{\epsilon_f + 2\epsilon_m - f(\epsilon_f - \epsilon_m)} \right)$$

- MG + DHM: Multiple reflections, phase delays, and wave guide modes are not considered
- **Absorption loss increases explosively**
- S-parameters: Experimental errors, antenna beam patterns, and other real-world effects
- **Slow changes between 8-13 dB**

01 INTRODUCTION

02 RESEARCH ACHIEVEMENTS

03 FUTURE PLAN



04 SUMMARY

05 APPENDIX



Composite Manufacturing

Smart optimization

GF RTP Manufacture

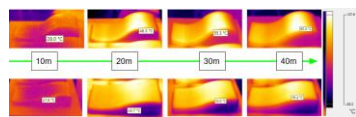
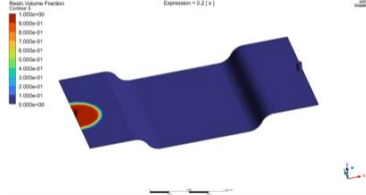
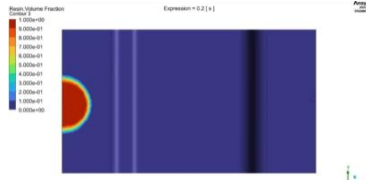
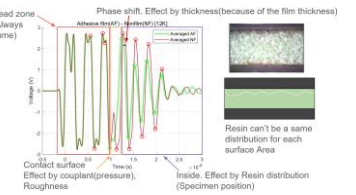
Comparative analysis of GF RTP manufacturing under different temperature, pressure, time, stacking angle, and vacuum conditions.

Microstructure Analysis

Cross-sectional analysis of the GF RTPs using Python.

ANN Prediction

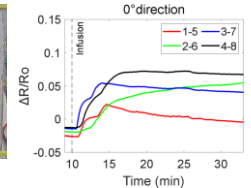
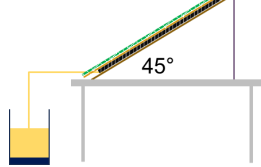
To develop ANN to predict the optimal conditions of GF RTP manufacturing.



Smart Composites

Functionalization

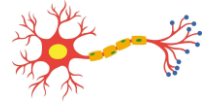
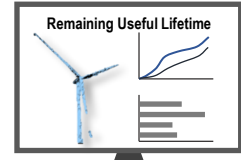
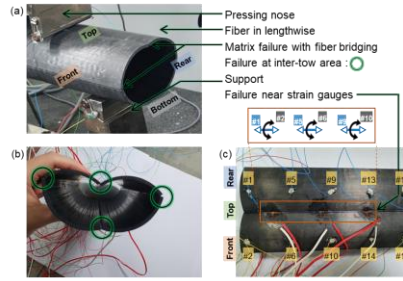
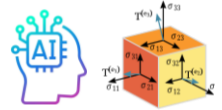
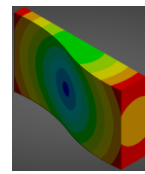
$$\frac{v^2}{2} + gz + \frac{p}{\rho} = \text{constant}$$



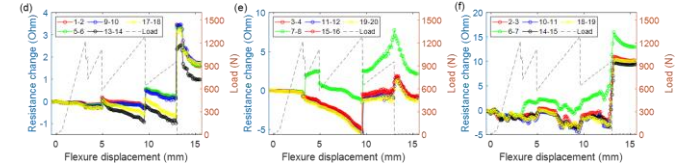
$$\begin{bmatrix} N_x \\ N_y \\ N_{xy} \\ M_x \\ M_y \\ M_{xy} \end{bmatrix} = \begin{bmatrix} A_{11} & A_{12} & A_{16} & B_{11} & B_{12} & B_{16} \\ A_{12} & A_{22} & A_{26} & B_{12} & B_{22} & B_{26} \\ A_{16} & A_{26} & A_{66} & B_{16} & B_{26} & B_{66} \\ B_{11} & B_{12} & B_{16} & D_{11} & D_{12} & D_{16} \\ B_{12} & B_{22} & B_{26} & D_{12} & D_{22} & D_{26} \\ B_{16} & B_{26} & B_{66} & D_{16} & D_{26} & D_{66} \end{bmatrix} \begin{bmatrix} \epsilon_x - \alpha_x \Delta T \\ \epsilon_y - \alpha_y \Delta T \\ \gamma_{xy} - \alpha_{xy} \Delta T \\ \kappa_x \\ \kappa_y \\ \kappa_{xy} \end{bmatrix}$$

$$\epsilon = \frac{L_T - L_0}{L_0} \rightarrow \alpha_i = \frac{\epsilon_i}{\Delta T} = \frac{\Delta L_i}{L_{0,i} \Delta T} \rightarrow \begin{bmatrix} \epsilon_{xx} \\ \epsilon_{yy} \\ \epsilon_{zz} \end{bmatrix} = \begin{bmatrix} \alpha_x \\ \alpha_y \\ \alpha_z \end{bmatrix} \Delta T$$

$$\{\epsilon\} = \{\epsilon^0\} + z \cdot \{\kappa\} \rightarrow \{\epsilon\} = \{\epsilon_T\} = \alpha \cdot \Delta T$$



Make Smart Composites, Intelligently



01 INTRODUCTION

02 RESEARCH ACHIEVEMENTS

03 FUTURE PLAN

04 SUMMARY



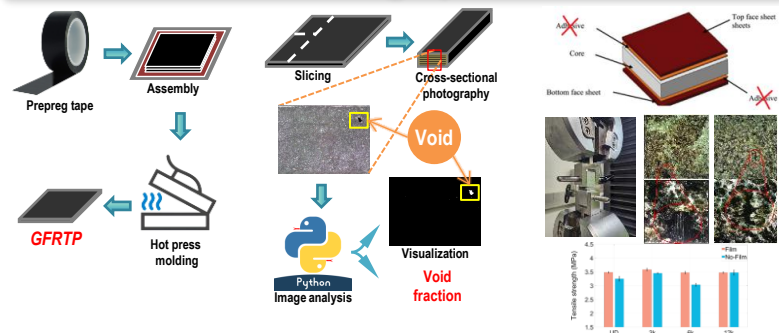
05 APPENDIX

Summary

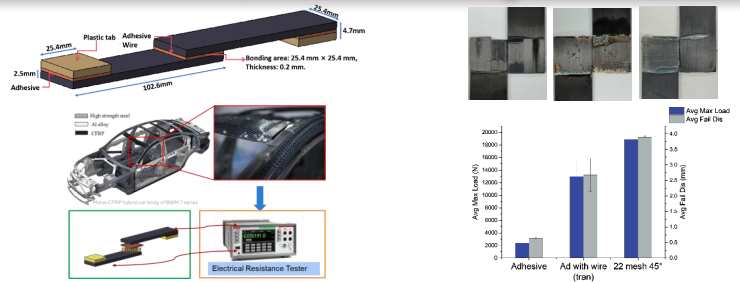
Manufacturing and Condition-based Structural Health Monitoring of CFRPs

Composite Manufacturing Optimizing & Joining

Manufacturing Optimization

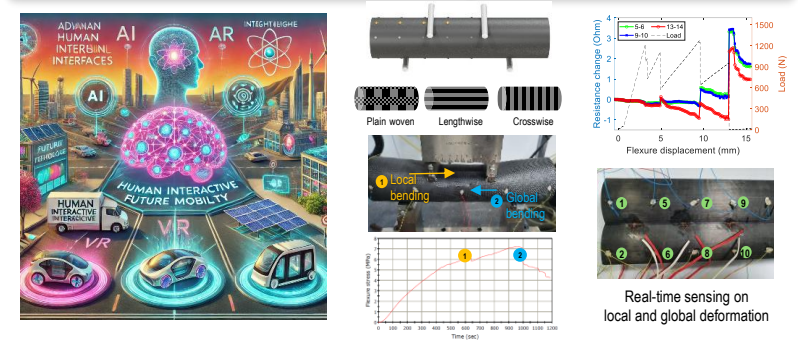


Fastener-free Joining

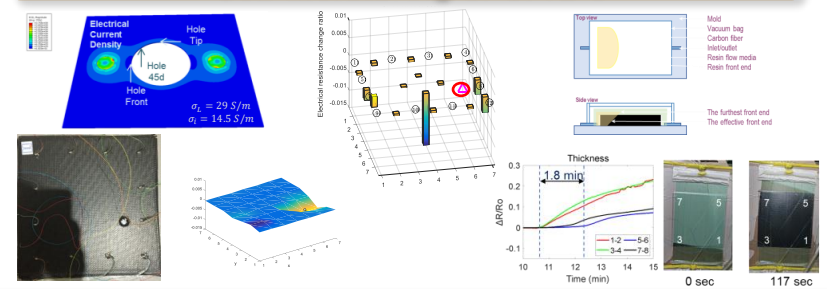


Smart Composites Functionalizing

Functional Composites



Self-monitoring of CFRPs



Make Smart Composites, Intelligently



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Thank you for listening

Make Smart Composites, Intelligently



*R*Lab



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Contents

01 INTRODUCTION

02 RESEARCH ACHIEVEMENTS

03 FUTURE PLAN

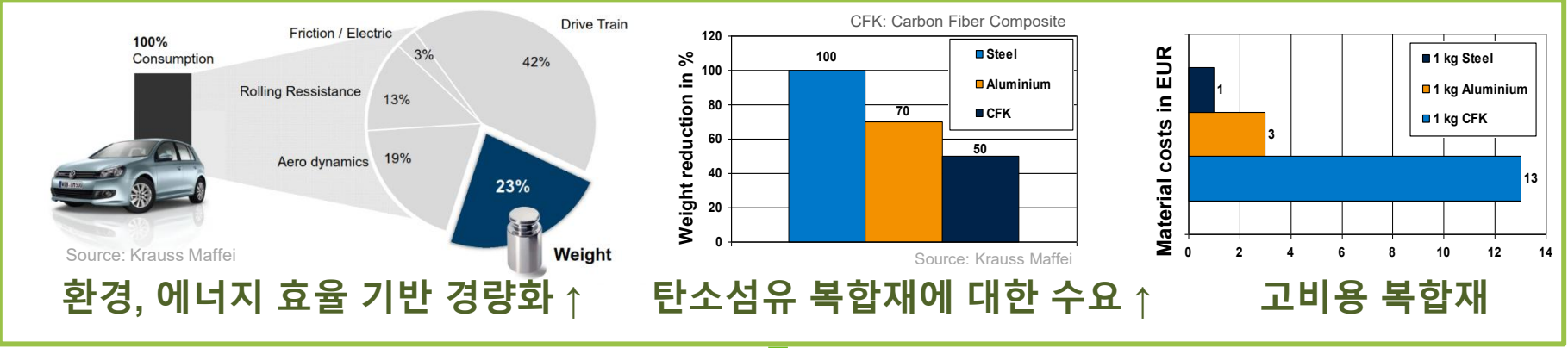
04 SUMMARY

05 APPENDIX



BeSTIS: Be Smart, Then It's Safe

Trend



환경, 에너지 효율 기반 경량화 ↑

탄소섬유 복합재에 대한 수요 ↑

고비용 복합재

Social Needs

- 복합재 채택 불가피
- 고비용 복합재 구조물의 구조 건전성 평가 필요
- 복합재 유지 및 보수 기술 필요
- 복합재 End-product 의 임무 수행 평가 필요

BeSTIS (베스티스)
Be Smart, Then It's Safe

Founder's Motto
이 세상을 이롭게 하자
편하게 살자
행복하게 살자

Composite



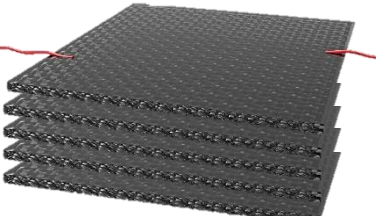
탄소섬유강화 플라스틱
(Carbon fiber reinforced plastic)



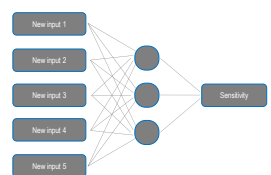
BeSTIS: Be Smart, Then It's Safe

탄소섬유 복합재의 구조 건전성 진단, 고장 예지 및 관리, 설계 → 종합 솔루션 제공

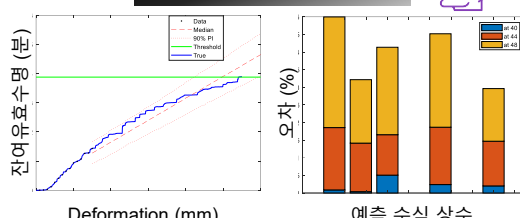
전극이 설치된 CFRP



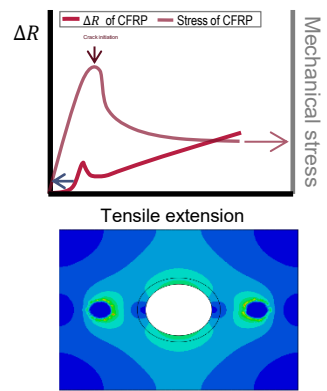
설계 피드백




고장 예지 및 관리




실시간 파손 진단과 물성 갱신

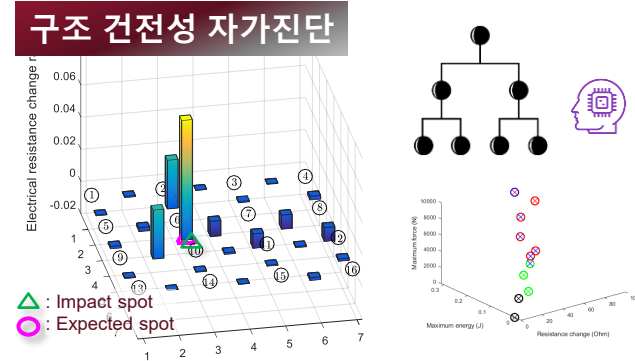




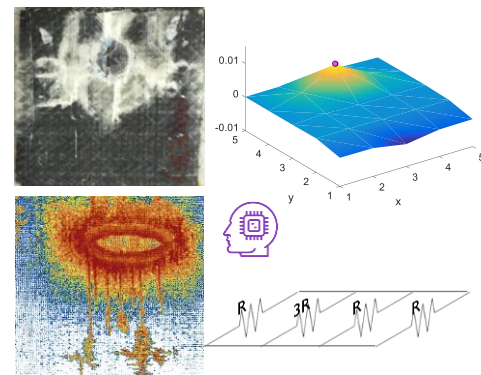
복수 개의 충격 파손



구조 건전성 자가진단



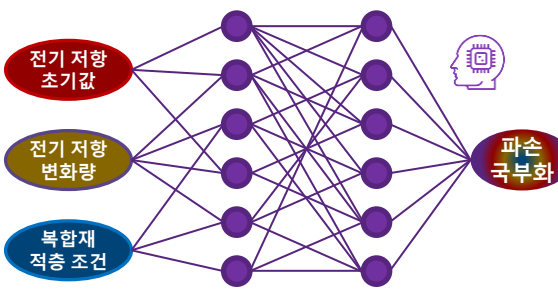
전기등가회로와 비파괴 검사



전기 저항 초기값

전기 저항 변화량

복합재 적층 조건

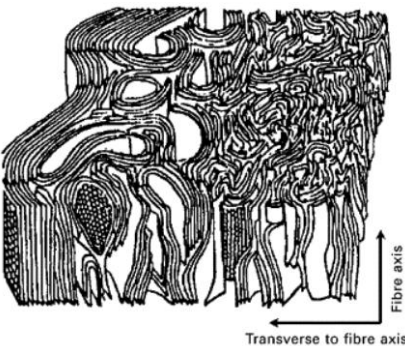


파손 국부화

Literature Review (Self-sensing)

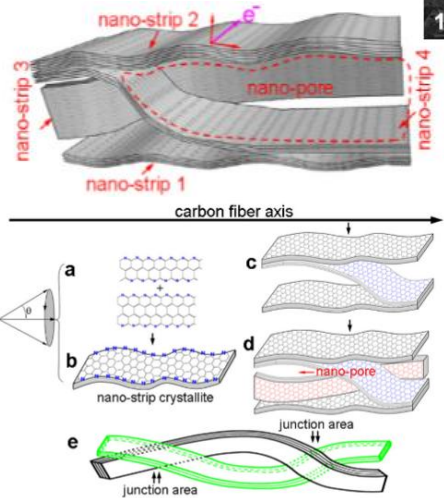
Carbon Fiber

Ref. #1

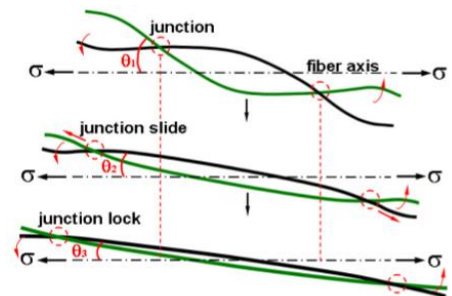
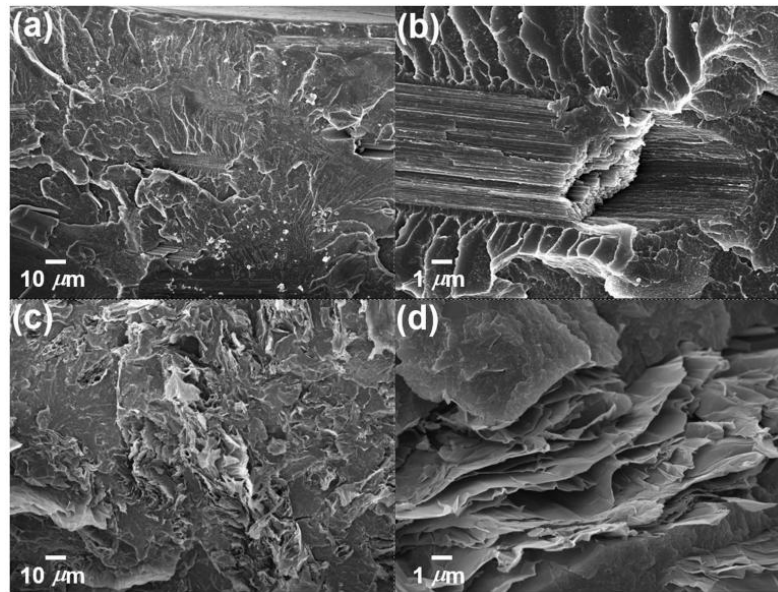


Carbon fiber = Graphite layers

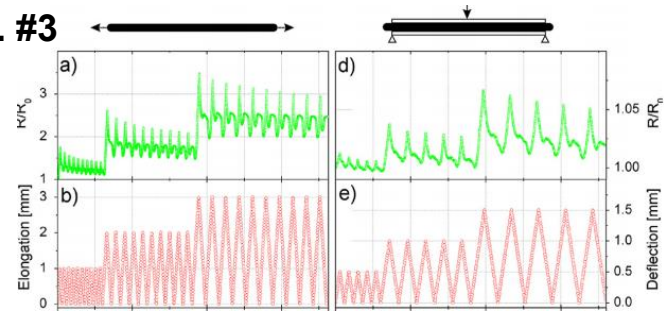
Ref. #4



Ref. #2

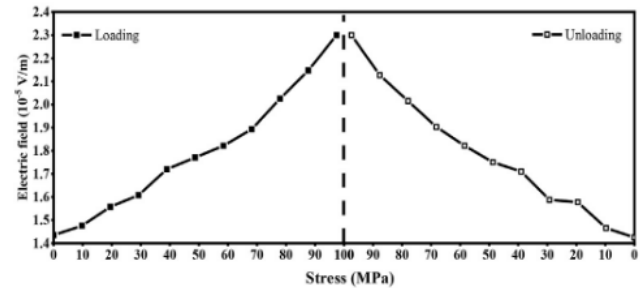
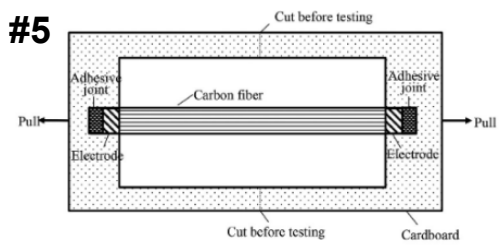


Ref. #3



Change in a graphite crystallite due to mechanical loading changes electrical resistance

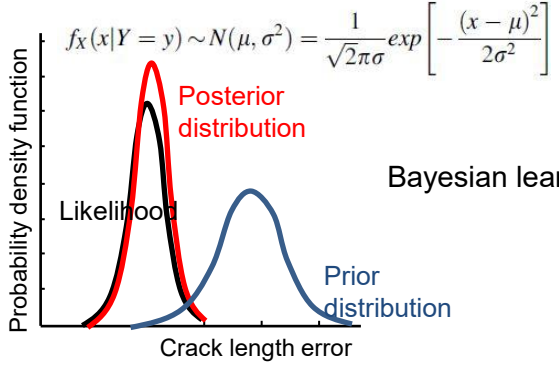
Ref. #5



#1. S. C. Bennett and D. J. Johnston. Proceed. 5th London Carbon and Graphite Confer. 1 (1978) 377-86
 #2. M. Lee *et al.* Scientific Reports. 5 (2015)11707
 #3. S. Cravanzola *et al.* Carbon 62 (2013) 270-77
 #4. G. Zhou, J. -H. Byun *et al.* Carbon. 76 (2014) 232-39
 #5. X. Xi and D. D. L. Chung. Carbon. 145 (2019) 452-61

Machine Learning

Machine Learning for Self-sensing

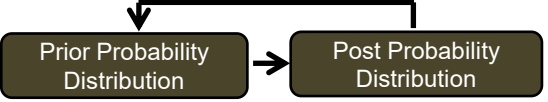


X : Uncertainty variable
 Y : Measured data

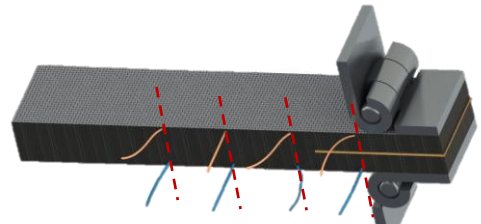
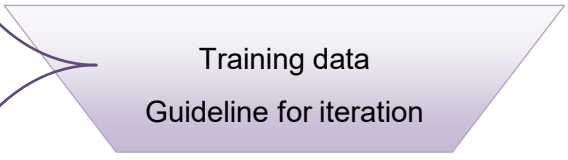
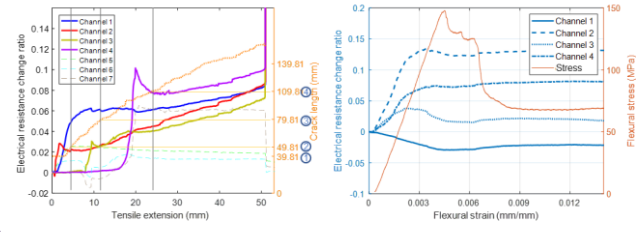
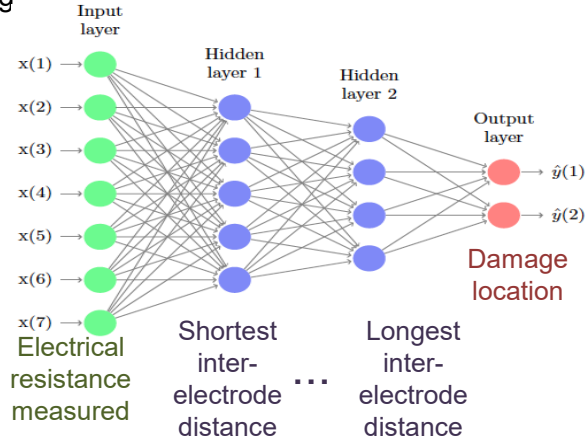
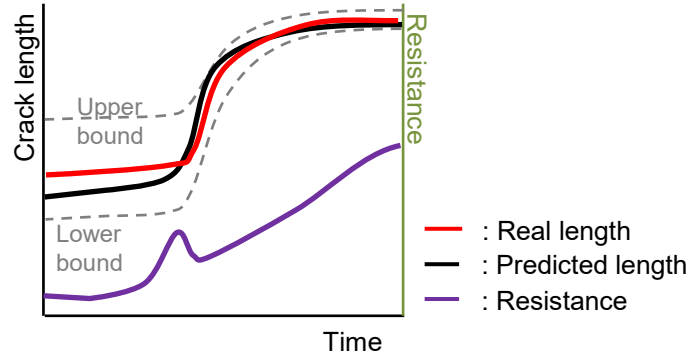
$$f_X(x|Y = y) = \frac{f_Y(y|X = x)f_X(x)}{f_Y(y)}$$

$$f_Y(y|X = x) = N(y, \sigma_y^2)$$

$$f_X(x) = N(\mu_0, \sigma_0^2)$$



$$p(\mathbf{D}|\mathbf{w}) = \prod_{i=1}^N p(y_i|\mathbf{x}_i, \mathbf{w}) = \prod_{i=1}^N (2\pi/b)^{-1/2} \exp\left[-\frac{b}{2}\{y_i - f(\mathbf{x}_i, \mathbf{w})\}^2\right]$$



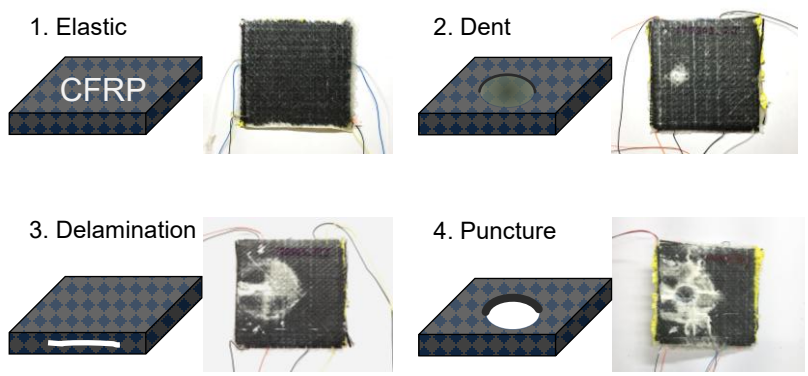
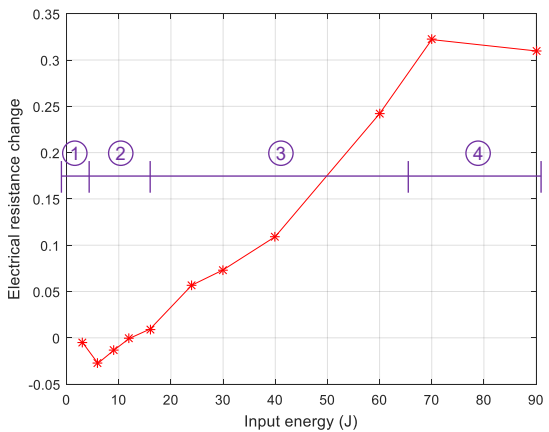
$$C(w, b) \equiv \frac{1}{2n} \sum_x \|y(x) - a\|^2$$

Cost function in backpropagation



AI (Artificial Intelligence) can simplify the self-sensing algorithm.

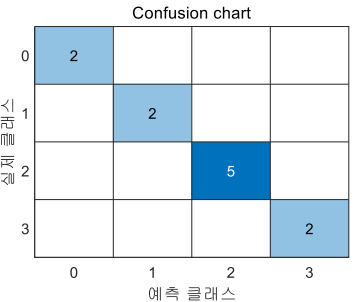
Machine Learning



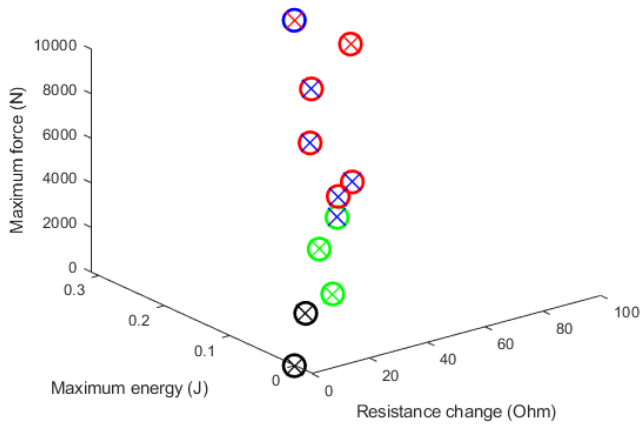
Damage Severity Analysis by Classification

CFRP: Carbon Fiber Reinforced Plastic
ML: Machine Learning
R: Electrical Resistance
E: Energy
F: Force

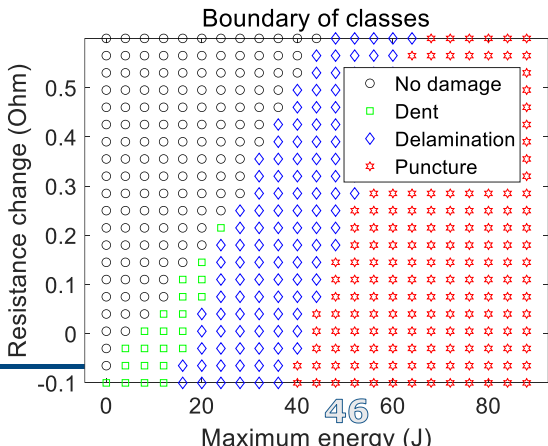
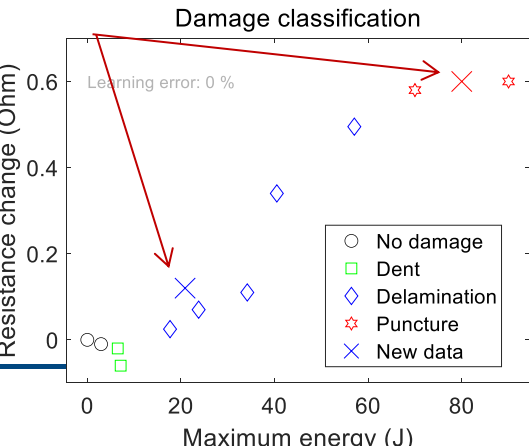
Actual data: Damage severity analysis of CFRPs



Input: Experimental data (R and E)
Tool: Classification by MATLAB
Output: Discriminate classes



ML: Classification of damage severity



Advanced ML: Fuzzy Clustering Method (FCM)

Input: Experimental data (R, E, F)
Tool: FCM by MATLAB
Output: Discriminate classes